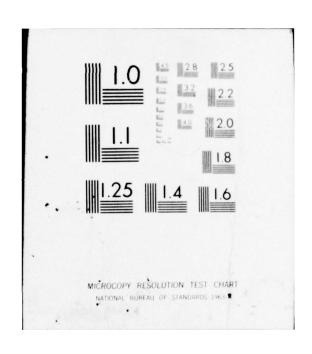
NAVAL POSTGRADUATE SCHOOL MONTEREY CALIF
AN OPTIMIZATION STUDY OF A LOW THERMAL POTENTIAL POWER SYSTEM. (U)
SEP 76 J R BUCKINGHAM, W M RAIKE
NPS-69KK76091
NL AD-A031 709 UNCLASSIFIED NL 1 OF 2 AD A031709





NAVAL POSTGRADUATE SCHOOL Monterey, California



COPY AVAILABLE TO DDC DOES NOT PERMIT FULLY LEGIBLE PRODUCTION



AN OPTIMIZATION STUDY OF A LOW THERMAL POTENTIAL POWER SYSTEM

> J. R. Buckingham W. M. Raike M. D. Kelleher

> > September 1976

DISTRIBUTION STATEMENT A
Approved for public release:

Approved for public release; Distribution Unlimited

NAVAL POSTGRADUATE SCHOOL Monterey, California

Rear Admiral Isham Linder Superintendent J. R. Borsting Provost

THERMOECONOMIC ANALYSIS OF VAPOR POWER SYSTEMS

A power generating system using the low thermal potential available from the vertical temperature distribution of the ocean is analyzed as a combined engineering and economic mathematical model. The model is optimized for minimum capital cost employing a sequential unconstrained minimization algorithm. Examples of the kinds of engineering and cost information available from the model are presented.

The work reported herein has been supported by the Energy Programs Office, Code L80, of the Civil Engineering Laboratory, Port Hueneme, California; work request N68305 75 WR-S-0068.

Matthew Kelleher
Associate Professor of

Associate Professor of Mechanical Engineering

Approved by:

Allen E. Fuhs, Chairman

Department of

Mechanical Engineering

Robert R. Fossum Dean of Research

NPS-69Kk76091 September 1976

Supersedes AD A028505
per NPS, Debbie Mc Howar,

8 nov X

UNCLASSIFIED

SECURITY CLASSIFICATION OF THIS PAGE (When Date Ent

(9) Final rept.

REPORT DOCUMENTATION PAGE	READ INSTRUCTIONS BEFORE COMPLETING FOR
NPS-69Kk76091	O. 3. RECIPIENT'S CAYALOG NUMBER
TIZLE form Subtitle)	S. TYPE OF REPORT & PERIOD COVI
An Optimization Study of a Low Thermal	1
Potential Power System	Final, FY76
	6. PERFORMING ORG. REPORT HUMB
AUTHOR(a),	8. CONTRACT OR GRANT HUMBER(e)
J. R. Buckingham, W. M. Raike, M. D. Kelleher	
PERFORMING ORGANIZATION NAME AND ADDRESS	16. PROGRAM ELEMENT, PROJECT, Y
Naval Postgraduate School	
Monterey, California 93940	N68305-75 WR-5-0068
. CONTROLLING OFFICE NAME AND ADDRESS	12. REPORT DATE 1 September 1976
Civil Engineering Laboratory	
Naval Construction Battalion Center	111 O PAGES
Port Hueneme, CA 93043 MONITORING AGENCY NAME & ADDRESS(If different from Controlling Office)	
	Unclassified
(W Sep 76)	To see an extension
	15a. DECLASSIFICATION/DOWNGRAD
Approved for public release; distribution un	limited.
Approved for public release; distribution un	
Approved for public release; distribution un	
Approved for public release; distribution un	
Approved for public release; distribution un	
Approved for public release; distribution units of the abetract entered in Block 20, If different in Supplementary notes	hour Report)
Approved for public release; distribution units of the abeliant antered in Block 20, if different in Supplementary notes 6. Supplementary notes	him Report)
Approved for public release; distribution units of the shetrest entered in Block 20, 11 different in the shetrest entered in Block 20, 11 different in the sheet sheet and in Block 20, 11 different in the sheet sheet and in the sheet sheet and in the sheet sh	ean thermal energy convers
Approved for public release; distribution units. 17. DISTRIBUTION STATEMENT (of the abstract entered in Block 20, 11 different in the state of the abstract entered in Block 20, 11 different in the state of the abstract entered in Block 20, 11 different in the state of the abstract entered in Block 20, 11 different in the state of the abstract entered in Block 20, 11 different in the state of the abstract entered in Block 20, 11 different in the state of the abstract entered in Block 20, 11 different in the state of the abstract entered in Block 20, 11 different in the abstract entered in the abstract entered in Block 20, 11 different in	ean thermal energy convers:
Approved for public release; distribution units of the shelfest entered in Block 20, 11 different in Supplementary notes 15. KEY WORDS (Continue on reverse side 11 recessary and identify by block much Nonlinear programming, thermodynamic cycle, occ	ean thermal energy convers
Approved for public release; distribution units of the abstract entered in Block 20, if different in the supplementary notes 18. Supplementary notes Nonlinear programming, thermodynamic cycle, occupy the supplementary optimization, cost estimation, system es	ean thermal energy convers
Approved for public release; distribution units of the shotrest entered in Block 20, if different in the shot of the shotrest entered in Block 20, if different in the short state of the short entered in Block 20, if different in the short entered in Block 20,	ean thermal energy convers
Approved for public release; distribution units 7. DISTRIBUTION STATEMENT (of the abetract entered in Block 20, if different in Supplementary notes 8. Supplementary notes Nonlinear programming, thermodynamic cycle, occupy the optimization, cost estimation, system entered in supplementary notes are reversed and if necessary and identify by block number of course of the information of the of the inform	ean thermal energy conversing ineering.
Approved for public release; distribution units 7. DISTRIBUTION STATEMENT (of the abeliant entered in Block 20, if different in Block 20, if differ	ean thermal energy conversing ineering. Thermal potential available the ocean is analyzed as a
Approved for public release; distribution units 7. DISTRIBUTION STATEMENT (of the abstract entered in Block 20, 11 different of the superior of the abstract entered in Block 20, 11 different of the superior of the abstract of the superior of the superio	ean thermal energy conversing ineering. Thermal potential available the ocean is analyzed as a model. The model is opti
Approved for public release; distribution units 7. DISTRIBUTION STATEMENT (of the abstract entered in Block 20, 11 different of the superior of the abstract entered in Block 20, 11 different of the superior of the abstract of the superior of the superio	ean thermal energy conversing ineering. The model is optimized as a model. The model is optimizential unconstrained minimum of the model is optimized to the model is optimiz
Approved for public release; distribution units 7. DISTRIBUTION STATEMENT (of the abstract entered in Block 30, if afferent in Supplementary notes 8. Supplementary notes Nonlinear programming, thermodynamic cycle, on system optimization, cost estimation, system estimation, system estimation of combined engineering and economic mathematical mized for minimum capital cost employing a sequence of the kinds of the system algorithm. Examples of the kinds of the system of the system.	ean thermal energy conversing ineering. The model is optimized as a model. The model is optimizential unconstrained minimum of the model is optimized to the model is optimiz
Approved for public release; distribution units 7. DISTRIBUTION STATEMENT (of the abeliant entered in Block 20, il different in the superior of the abeliant entered in Block 20, il different in the superior of the abeliant in the superior of the abeliant in the superior of the superio	ean thermal energy conversing ineering. The model is optimized as a model. The model is optimizential unconstrained minimum of the model is optimized to the model is optimiz

DD 1 JAN 73 1473

E/H 0102-014-6001 |

UNCLASSIATED
SECURITY CLASSIFICATION OF THIS PAGE (Miss Date Miss.)

AESTRACT

SANDARD A TO DOME TO SAVE I

REPORT DOCUMENTATION PAGE

EARLY NI HIS DEED TO THE THE PARTY TO SEE THE

W. RE: WORLD C. College or respective all autospices and respect to block months.

Canana cand of allered has reserved to the or was expected by These tens.

to meson our lobus att mort shallows solven

Staroff of the Brand Worth Colored Compactors of P

THE WAY TO THE

SUR ON ARM LAND WAS TO US TO

A power generating system using the low thermal potential available from the vertical temperature distribution of the ocean is analyzed as a combined engineering and economic mathematical model. The model is optimized for minimum capital cost employing a sequential unconstrained minimization algorithm. Examples of the kinds of engineering and cost information available from the model are presented.

Wallingto exceedancy, the metadamic center, calcan sterces energy concerning,

oldr izen telmet o ligned mit mit mit heter indep anligeben opein h

Committee of the party of the state of the s

A ser to relate at nervo set to collective the market gast inputer out over - item of the beautiful of the second of the second

TABLE OF CONTENTS

I.	IKI	ECITICAL	12
	A.	EJCKGRCUND	12
	В.	CEJECTIVES	20
II.	TEE	MCCEL	21
	٨.	GENERAL DESCRIPTION	21
	E.	CYCLE ANALYSIS	25
	c.	ENGINEEFING FRAMEWORK	30
		1. The Bciler	30
		2. The Condenser	45
		3. The Pumps	47
	D.	CCS1 FRAMEWCRK	49
	E.	ASSEMBLY OF THE MATHEMATICAL MODEL	53
		1. Objective Function and the Constraints	55
		2. the optimization method	57
III.	RES	DITS AND CONCLUSIONS	62
IV.	REC	CHPENIATIONS	71
tteng	ix !	CCMPUTER PROGRAM AND SAMPLE DATA	73
IST O	F FE	FIFINCES	106
AITIK	I EI	STRIBUTION LIST	109

ACCESSION	14			-
HTIS		White Sec	ilea	M
980		Buff Secti		
BMANAGUN	CEO			0
JUSTIFICAT	10H			
Diet.		and/or		
^				
	1	A CONTRACTOR OF THE PARTY OF TH		
N				



Approved for public release;
Distribution Unlimited

LIST OF TAPLES

1.	Material Factors for the Heat Exchangers	51
2.	Material Factors for the Centrifugal Pumps	51
3.	Material Factors for the Propeller Pumps	52
	Defirition of the Blements of the Vector X	54
5.	Problem Input Data	64
6.	Example Froblem Solution Summary	65
7.	Solution Fower Budget	68
8.	Additional Data from Sample Problem Solution	70

LIST OF PIGURES

1.	System Schematic Diagram	16
2.	System Arrangement	22
з.	Theracdynamic Cycle	27
4.	Bciler Crcss Section	31
5.	Comparison of a Single Tube Versus	
	a Tuke in a Bundle	37
6.	Thermal Network for a Single	39
7.	Problem Sclution Sequence	59
8.	Cost Ereakdowr	67

LIST OF SYMPOLS

UNITS

SYMBOLS DESCRIPTION

SIEEUL.	LESCAIPTICN	UNIIS		
		ENGLISH	METRIC	
1	area	in ²	CM2	
c	heat caracity	BTU/1bm-oF	J/kg-cc	
c	capacity rate	BTU/hr-05	kW/°C	
đ	tuke diameter	in	CB	
Cs.	diameter of the shell	ft	n	
f	Fanning friction factor	tes arrandesen	1-(8) -5	
F	factor for the tube		-	
	bundle toiling	tecopeasic Oyc		
9	inequality constraints	Mag addio tel	J=19 . P.	
h	equality constraints		-	
h	heat transfer coefficient	BTU/ft2-hr-oF	kW/m2-0C	
В	erthalpy	BTU/1bm	J/kg	
В.	erthalpy corrected for	BTU/lbm	J/kg	
	sutcocling			
hs	height from midplane of	ft ft	014 - /	
	the tciler to top of	TRODALDED 3	960 .6	
	the tute bank			
i	varcr quality	-	-	
k	thermal conductivity	BTU/hr-ft-oF	k W/#-0C	
Kc	pressure loss factor at	-	-	
	the tube entrance			
Ke	pressure loss factor at	-	•	
	the tube exit			
I	tule length	ft	•	
E	mass flow rate	lbm/hr	kg/hr	
n	average number of tubes in	-	- 10	
	a cclren in the condenser			
Neg	specific speed of a pump	RPM	RPH	
NIU	number of heat transfer	•	-	
	units of an exchanger			
8 u	Nusselt number	-	•	
F	pressure	lbf/in2	kp	

F	power	BTU/hr	RM
Fi	tuke pitch	in	CB
Fr	Pranôtl number	- 30/100	-
9	heat flux	BTU/hr-ft2	k4/#2
C	heat transfer rate	BTU/hr	Ha
Fn	heat transfer	hr-op/BTU	oc/kw
	resistance (R1,R2,R3,R4)		
F€	Reynclds number	- 5100 FLF	-
8	entropy	BTU/1bs-05	kW-hr/kg-OK
S	number of shells	■Ly Crayer, At	-
51	segmental area	ft ²	m 2
t	tute wall thickness	in	CM
1	terperature	07-7-12-14-	°C
u	velocity	ft/sec	m/sec
0	overall heat transfer	BTU/ft2-hr-or	k W/m2-0C
	ccefficient		
	specific volume	ft3/lbm	m³/kg
VI	varct lcad	lbm/hr-ft3	kg/hr-m3
y	propertionality factor in	■ \$1258 383 =	-
	the transition region		
1	capital cost	dollars	dollars
REEK			
STREOL	.s		
E	heat transfer effectiveness	wosar.alle	•
7	efficiency		
λ	proportionality factor in	-1200795545	•
	the transition region		
-	dynamic viscosity	lbm/sec-ft	N/sec-m
y	kinematic viscosity	ft2/sec	m2/sec
•	density	1bm/ft3	kg/83
σ	surface tension	lbf/ft	N/B
•	free flow/frontal area	-entry and	1.
	ratio		

SUESCRIPTS

- ab abscried
- t tciler
- tulk average of the inlet and outlet temperatures
- c condenser
- c critical pressure
- the cclder fluid in the heat exchanger
- car carnot efficiency
- cp circulation pump
- cyc theracdynamic cycle
- f saturated liquid
- fg change in the quantity as

 the fluid changes from all

 liquic to all vapor
- fp feed pump
 - g saturated vapor
 - the heat exchanger
- he heat exchanger
 - i inside
 - in fluid entering the exchanger
 - I lawinar flow regime
 - s sean temperature difference
 - c max. possible temperature
 difference
 - c outside surface
 - cut fluid leaving the exchanger
 - F, at constant pressure installed the state of the
 - rei rejected
 - sw seawater
 - sys the system was acreed the said
 - Ir transition regime
 - Tu turtulent regise

rarest english

B BOOK TE

turtine wall wf working fluid C maximum temperature difference 1 state point 1 2 2 25 25 3 3 5 The bodies aga . First in the And the Area Ser

11

I. INTRODUCTION

A. EACKGECUND

As cf 1974, the United States contained only 6% cf the world population but was using 33% of the energy consumed each year according to Ref. 1. Approximately one third of the oil used in the U.S. is imported. This heavy dependence on foreign scurces for energy coupled with the exhaustion of dcmestic fessile fuels, particularly oil, in the foreseeable future is causing the U.S. military and civilian sectors of the economy to search for energy sources that would be inexhaustible and independent of fcreign ccntrol. Increasing fuel costs and environmental problems are other forces fushing the search for alternatives. Nuclear power, once considered the cure-all, is facing severe problems. Costs of construction, operation, and fuel have risen dramatically. Questions concerning the safety of operation of the nuclear plants and of the storage of nuclear wastes have become political issues that have exploitation of nuclear power. The present nuclear reactors use a fuel that, like petroleum, has a limited availability and whose price is increasing rapidly. The breeder reactor, which would create more fuel than it burns, and the fusion reactor, which would use the hydrogen found in water, are still a long way from producing power for a world whose demands for energy increase each year.

Many pecile advocate the exploitation of the so-called "free" energy sources. These sources are: (1) the kinetic

energy of moving fluids, such as winds and coean currents;

(2) the potential energy of tides and rivers; (3) the heat generated inside the earth; (4) the direct conversion of solar energy into electricity and heat; and (5) the use of sun warmed water in conjunction with a source of colder water to provide the temperature differential to run a vapor power cycle. None of these sources are truly "free". What is happening is the trading off of the transportation, processing, and environmental costs of a concentrated energy source for the capital, operating, and social costs of converting a very diffuse source into a more concentrated form such as electricity.

The earth can be thought of as a giant heat engine, abscrbing energy from the sun and reflecting energy as thermal radiation tack to space as pictured in Ref. 2. Since the poles receive less energy than the equator, the atmosphere and ocean attempt to distribute the energy more evenly ever the earth. The air is heated by absorption of sunlight and ty contact with the surface of the earth. heated air rises thus setting up surface wind currents as cccler air tries to replace the rising air. stress created by the relative motion between the air and the surface water causes the water to move thus creating the surface water currents. Hany currents in the oceans, regardless of depth, originate with the wind shear stress at the air-surface interface. The circulation patterns of the oceans become extremely complex due to the influence of the rotation of the earth and the shapes of the ocean basins.

In general the surface waters near the equator are warmed by the sun and flow towards the poles giving up energy by evaporation, and by convection to the air and by radiation back to space. As these currents cool, some of the water becomes dense enough to sink and then flows back to the equator along the bottom.

The relationship between salinity and temperature and the density of seawater is such as to create a stable stratification of the ocean in many areas. A salinity decrease or a temperature increase reduces the density. The total effect is to create a warm surface current overlaying the cold water returning from the poles. In the tropical waters the surface temperature may exceed 80°F (26.7°C) while 3,000 to 6,000 ft. (914 to 1828 m) down the temperature may be about 40°F (4.4°C). This temperature difference can be used to rur a man-made heat engine, like the Ocean Thermal Energy Converion (OTEC) power plant concept.

The proper choice of a site is critical to the cost of such a heat engine. Since the temperature difference is low compared to conventional thermal power generating methods which typically use temperature differentials exceeding 500°F (260°C), the effect of a 1°F (0.56°C) loss in the AT is much more costly at the low thermal differentials, which are usually less than 50°F (27.8°C) in the Therefore, the vertical temperature profile of the cceans will centrel the location of an OTEC power plant. surface temperatures vary with location and with time. At most places in the ocean there is a seasonal change of the surface water temperature of 5 to 8°F (2.8 to 4.4°C) except in the tropical areas that remain relatively constant according to Ref. 3. However, all regions of the oceans are affected by wind-generated mixing of the surface water that changes the surface temperature by a degree or so and a daily cycle cf a few tenths cf a degree. Some of the possible U.S. near-shore sites being considered are the waters around the Eawaian Islands and the Gulf Stream off the southeastern U.S.

In the past, the engineering analysis and the cost estimation have been conducted as if they were separate

furctions. In a system with a small temperature differential, such as this one, the engineering design and economic analysis should be linked together as a single procedure because of the capital intensive nature of the ocean thermal power plant. For example, the heat transfer equation (C = UA AT) is coupled to the cost equation for heat exchangers ($\$ = K A^n$) through the surface area of the tutes. Eccause of the small AT involved the area must be very large to supply the same Q. The other components are all linked together by the engineering and cost equations so that the fiction cannot logically be broken into equipment ty equipment optimization. The present analysis considers a closed wapor power system operating on the thermal rotential available from the ocean. The economic optimization of the system is carried out to show what information can be gotten, what conclusions can be reached fice such analysis and where research effort should be expended to improve the design of the system. This study does not attempt to predict the cost of such a system or to make specific recommendations about the parameters.

A schematic diagram of the system is shown in Figure 1. The warm surface water, the Gulf Stream for example, is pumped through a heat exchanger. There some energy is transferred by boiling the working fluid such as ammonia. The working fluid is piped to a turbine where some of the energy is converted to mechanical energy then to electrical energy is a generator. The unavailable portion of the absorbed energy is rejected to the environment in the condenser and the working fluid is returned to the liquid state. The low temperature in the condenser is maintained by pumping cold ocean water up from the depths. Finally the working fluid is pumped back to the boiler.

The Coear Thermal Energy Conversion program has received much attention in the last several years from many

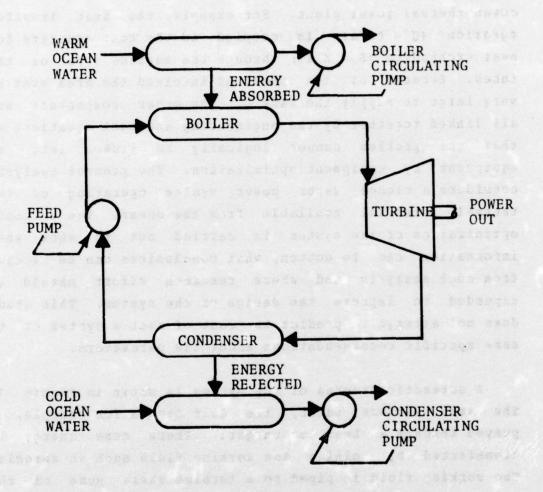


FIGURE 1. SYSTEM SCHEMATIC DIAGRAM

researchers funded at first primarily by the National Science Fourdation (NSF). Since the establishment of the Bnergy Research and Development Administration (ERDA), new research contracts are funded mainly by this agency. The U.S. Navy is also funding some research in the OTEC area. Ref. 4, 5 and 6 give a good summary of the work that has been done. While much of the work has concentrated on the heat exchangers, the other components and the system as a whole are receiving attention. Some researchers are concentrating on the environmental and legal questions of such systems. Other teams are considering how to use the energy surplied from an OTEC power plant.

In any design, trade-offs must be made. Capital cost can be traded for operational costs. Pumping power can be traded for heat exchanger size. Trade-offs can involve items that have a common measure such as dollars; some other trade-offs are very hard to measure in common terms such as social and environmental costs. The trade-offs may involve only one component such as a pump or a turbine, but most often the trade-offs affect more than one component or possibly the entire system.

The trade-cff process should continue until an optimum design is obtained. From Ref. 7, "The aim of optimization is the selection, out of the multiplicity of potential solutions, of that solution which is the best with respect to some well-defined criterion." In order for the plant design to be optimum, the components must be optimized within the context of the entire system. Some of the parameters for a piece of equipment may have no direct effect on the optimum design of another portion of the system, but some parameters will have significant effects. For example, a heat exchanger is made up of a large number of tubes of a given diameter and length and is designed to have a certain heat transfer rate for a given flow rate of

water pumped through the tubes. The diameters, length and the number of tubes may be optimized for lowest cost of the heat exchanger. This approach leaves out the effect that the tube diameter and the tube length have on the pressure drop across the heat exchanger. Minimizing the cost of the heat exchanger involves minimizing the surface area. This may lead to increasing the pressure drop across the exchanger. The higher the pressure drop the more powerful and the more expensive the pump must be to maintain the flow velocity. This is just one example of the hundreds of trade-offs possible.

Unforturately, what happens in many cases is that each group of experts working on a particular component optimize that item around some given conditions that may have been set by teams working on some other section of the system. Little analysis is made of how the entire system responds to the coupling constraints. All that the pump manufacturer wants is that the buyer tell him what the flow rate, head and service conditions are and he will guote a price. Before the buyer can decide on the flow rate and head he should first know how the cost of the pump varies as flow rate and head vary so that those two parameters can be optimized in the context of the system.

The problem then becomes one of acquiring sufficient information on how costs vary as certain parameters change. Each manufacturer is able to make estimates of costs for his product but he is reluctant to devulge his information to buyers or competitors. This makes constructing cost curves a hazardous process at best, since the data must be gathered by other methods that may cause large unexplained variances.

Sometimes the optimization analysis is made with assumptions that make the answers of questionable value. For example, sometimes the heat transfer coefficient, U, is

assumed fixed in the equation q = UA AT. But, as shown in Ref. 8, for holling the heat transfer coefficient is very sensitive to the temperature difference with the result that q = UA'AT^{3.33}. The The accuracy of the engineering equations should be kept in mind. Most of the equations are correlations to fit experimental data. The correlation was developed for some particular set of data and may disagree significantly when compared to data taken by other experimenters. The limitations of the various relations must be kept in mind when reviewing the results. For a system the size of the OTEC plant, a pilot plant should be constructed in order to prove out the answers given by the calculations before expensive mistakes are made.

As a design progresses, the analysis begins with a general ficture containing many simplifying assumptions and proceeds foreward with more details added at each stage. This report is the second one written at the Naval Postraduate School on the subject of OTEC. Commander Furman Sheppard, USN, in his analysis, Ref. 9, attempted to show what kinds of information could be gained from a combined thermal economic model. He advocated the use of zone analysis, as developed by Wismer in Ref. 10, for large complex systems. In Wismer's approach to zone analysis, the system is broken down into zones containing one cr components. The zones are connected by linking variables. Two methods of optimization are possible. For the first method, in each zone, the zone parameters are optimized while the lirking variables are held fixed. Then, the zone parameters are held fixed while the linking variables are optimized. The process iterates between optimizing the zone parameters and the linking variables until convergence criteria are satisfied. In the second method the zones are cut off tree each other and each linking variable is separated into two variables, one on each side of the cutting place. The optimization proceeds until the linking variables on each side of the cuts are equal.

Sherrard's system did not produce vapor in a boiler but, instead, maintained the working fluid in the liquid state in the warm seawater heat exchanger and vaporized the working fluid, amuchia in his model, by expanding it through a "black box" called a vaporizer. His optimization analysis was confined to the zone containing the warm seawater exchanger, the feed pump, and the warm water circulating rump.

E. CEJECTIVES

The present analysis improves the previous model analysis by replacing the liquid-to-liquid heat exchanger and the vaporizer with a boiler. The boiler model is to be of sufficient detail so that a realistic design is represented. The other major components of the system, such as the turbire, the condenser, and the condenser circulating pumps, are included. A nonlinear programming technique is used to perform the thermal-economic optimization of the completed system. The research reported here is intended to show the usefulness of optimization analysis in the design of a low thermal difference system and to demonstrate some of the problems that must be considered in this type of analysis.

II. THE MODEL

A. GENERAL DESCRIPTION

Vapor power generating cycles usually consist of four processes. Energy is absorbed from a high temperature source and is used to vaporize a working fluid in a boiler. Next, work is extracted by expanding the vapor in an engine, exhausting the vapor at a lower temperature and pressure. The working fluid is then condensed back to the liquid state by rejecting energy to the lower temperature energy sink in the third process. Finally, the liquid is returned to the higher pressure of the boiler by a pump. In the following analysis, the energy source is a current of ocean water leated by the sun and the energy sink is assumed to be colder ocean water from deep currents. The working fluid is assumed to the propane although many other fluids are possible, such as ammonia or the Freens.

Six components are modeled in the following analysis. The hot side heat exchangers, the boilers, vaporize the working fluid which is expanded through a turbine exhausting to a conderser that rejects the unused heat to the cold sizk. Three rumps are included, a feed pump returns the working fluid to the boiler and two pumps circulate the warm and cold waters through the boiler and the condenser, respectively. The arrangement of the system is shown in Fig. 2. The nodes are numbered to correspond with the fluid state points shown later, in Fig. 3.

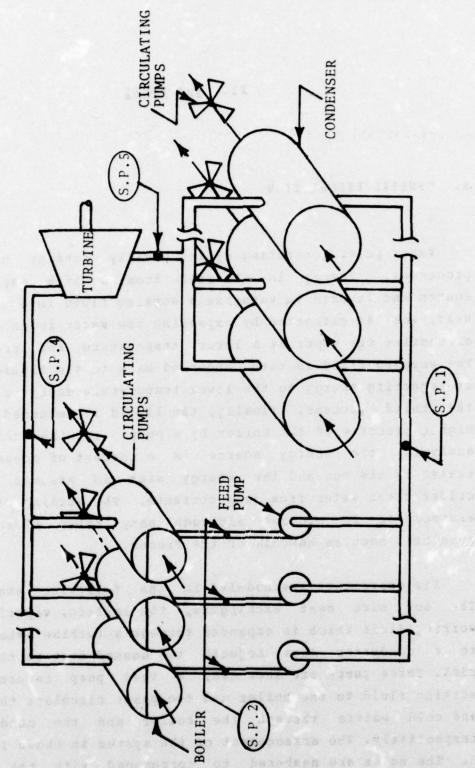


FIGURE 2. SYSTEM ARRANGEMENT

The analytical model consists of the equations describing the system and a set of parameters to be crtimized sc as to minimize the capital cost of the plant subject to certain constraints. The objective function, the function to be minimized, is the sum of the capital costs of the pumps, the boiler and the condenser. Although the turkine is a significant capital cost and would affect the design of the system, it is not included in this model. There are two classes of constraints. The first class is the explicit constraints that specify the upper and lower bounds on the parameters. The other class of constraints is the implicit constraints. These are the engineering equations that describe the operation of the power plant. example, a constraint might say that the temperature inside the boiler must be less than the temperature of the warm seawater because the equations fail if the requirement is nct satisfied. Another example would be a power balance that requires that the sum of the energy flows into and out of the system to zero. Additionally, they proscrite the region that the optimization routine may search within for the minimum of the problem. The implicit constraints are of two types, equality and inequality constraints. Inequality constraints set one-sided bounds on the feasible region. Scre of the inequality constraints are restrictions placed on the maximum or minimum values of the parameters due to judgements made on the basis of practical engineering considerations. The equality constraints are relationships that must be strictly satisfied at the optimimum. The system of equations describing the model fit into the following fcis:

minimize
$$f(X)$$

subject to: $g(X) \ge 0$, $i=1,2,...,m$
 $h(X) = 0$, $i=m+1,...,m+n$

where X is the vector of the parameters to be optimized, f(X) is the capital cost function, $g_i(X)$ are the inequality constraints and $h_i(X)$ are the equality constraints. The parameters are the dimensions of the boilers and the concensers, the boiling and condensing pressures, the sea water velocities, and the mass flow rate of the working fluid through the system.

The lasis for the economic framework must be set cut at the start. All measurable costs are relevant to the final design of the power plant, but, for some types of analysis and design, many costs can be ignored. The life cycle cost of the CIEC must be considered before the decision is made to commit the U.S. and the rest of the world to the consequences of choosing this energy generation scheme as even a partial solution to the energy problem. cycle cost consists of the capital costs, the financing costs, and the operating costs. The trade-cff of operating cost and capital cost is not considered in this research, because the operating cost is, for the independent of the overall dimensional characteristics of Operating costs are dominated by the the compenents. structure, site location, and maintenance requirements. Such things as the kind of instrumentation, controls, tearings, and structural materials have more effect on the operational costs than the length of the boiler or the diameter of the toiler shell. Therefore, only the capital costs of the major components, except the turbine, are considered.

To be useful for model analysis cost data must be transformed into a cost estimating relationship (CFR) that shows how cost varies with some variable or some combination of variables. The costs must be demonstrated to be highly correlated with the variable chosen. Since the relation is only a correlation, the bounds on the accuracy of the equation should be specified by those who develop it. Since the relation is arrived at from a finite body of data, the CER is directly applicable only in the range of the data Extrapolation of the CER outside the rarge is dargerous tecause there is no data in that region to support the assumed curve. However, the value of CER's is the use of them to enable the estimator to make a better analysis of a new design than he could if the information is not The relevant range of the CER must be kept in mind wher analyzing the solutions from the standpoint of the reliability of the answers. Another factor to consider when using cost estimating relationships is whether the equipment the data was taken from is comparable to the system the analyst is considering. If the design, the technology, or the application of the proposed equipment is significantly different fice that from which the data is taken, the CER may give unreliable answers.

This aralysis begins with the development of the engineering equations that constrain the system. This is followed by the development of a cost framework based on capital costs as the function to be minimized. The engineering and economic relations are linked together in the objective function and the constraints. A description of the algorithm that performs the optimization is included.

B. CYCLE ANALYSIS

The tuilding of the model begins with the analysis of the thermodynamic cycle. Pigure 3 is a temperature-entropy diagram of the working fluid on which the cycle is shown. These numbered nodes represent the thermodynamic state points at that location in the system and correspond to the numbers in Fig. 2. The boiling and condensing processes are assumed to be reversible processes. In the pump and the turbine the reversible process is indicated by the subscript "s" on the state point number. In the actual system there are irreversible losses inside each component that are left out of the analytical model or lumped in with other losses. The pressure drops due to frictional losses in the boiler and the condenser are neglected.

State point 1 assumes that the liquid leaving the toiler is saturated. State points 2s and 2 represent the conditions leaving the feed rump. If there were no losses in the rump, the fluid would be at 2s. The fluid enters the boiler to be heated, first, to saturation conditions at state point 3 then boiled at essentially constant pressure. State point 4 shows the vapor leaving the boiler as a saturated varor. The lines from point 4 to 5s and 5 represent the expansion through the turbine. The path to state point 5 accounts for the losses due to friction and other irreversible processes in the turbine. The remaining heat is then rejected from point 5 to point 1, again assumed to be a constant pressure process.

The theoretical heat transfer rate in the boiler and the condenser is found by subtracting the enthalpy at state points 2 and 1 from the enthalpy of state points 4 and 5, respectively, and multiplying by the total mass flow rate of the working fluid.

$$Q_{ab} = M_{wf}(H_{\psi} - H_{z}) \tag{1}$$

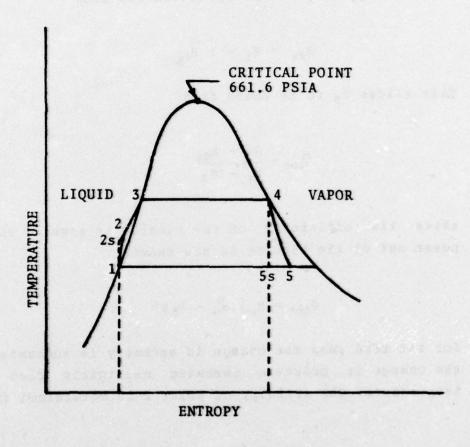


FIGURE 3. THERMODYNAMIC CYCLE

$$Q_{rej} = M_{wf}(H_5 - H_1)$$
 (2)

The enthalpy at state point 5 is determined by first finding the moisture content of the fluid exiting the turbine.

$$i = \frac{s_4 - s_1}{s_{fg}} \tag{3}$$

The enthalpy at point 5s is determined from

$$H_{5s} = H_1 + i H_{fq}$$
 (4)

This allows P5 to be found from

$$7_{tur} = \frac{H_4 - H_{5s}}{H_4 - H_5}$$
 (5)

where the efficiency of the turbine is assumed known. The power out of the turbine is now known.

$$P_{tur} = H_{wf}(H_{4} - H_{5}) \tag{6}$$

For the feed rump the change in enthalpy is accounted for by the change ir pressure assuming negligible rise in the temperature. The enthalpy of point 2 is determined from

$$\gamma_{fp} = \frac{v_1 (p_{2s} - p_1)}{H_2 - H_1} \tag{7}$$

where the efficiency is assumed known. The power required to drive the pump is determined from

$$P_{fp} = M_{wf}(H_2 - H_1)$$
 (3)

To gain a better feeling for the low thermal potential systems and to make comparisons with other systems several efficiencies are worthwhile calculating. The first is the Carnot efficiency which defines the maximum efficiency possible for a system operating between two temperature limits in degrees absolute, OR (OK).

$$7_{car} = 1 - \frac{T_{cold}}{T_{hot}}$$
 (9)

The next efficiency calculation takes into account the theoretical thermodynamic cycle and is the ratio of the net power out of the cycle to the heat taken up by the cycle

$$P_{cye} = \frac{P_{tur} - P_{fp}}{Q_{ab}} \tag{10}$$

This accounts for the thermodynamic inefficiencies inherent in the cycle. Finally, the system efficiency takes into account the other internal consumers of power: the circulation rumps, the pipe losses, and any other losses, such as the antifouling equipment. This calculation accounts for all of the mechanical and electrical inefficiercies of the system.

For example, if the hot water is at 75°F (23.9°C) and

the cold water is assumed at 40°F (4.4°C) the Carnot efficiency is 6.5%. For one typical set of assumptions, as shown in Table 5, where the turbine and the pumps are assumed to be 85% efficient, the cycle efficiency is 2.6% and the system efficiency is 2.4% for one feasible design.

C. ENGINEERING FRAMEWORK

This section describes the design of the components that make up the system. The level of detail of the analysis provides a physically realistic model and closes the loop. The pipe friction losses on the cold and hot water inlets as well as the pipe losses in the working fluid side are neglected. The turbine is included only as an energy transducer and no attempt is made to obtain costs, physical dimensions, or fluid flow characteristics. The efficiencies of the turbine and the pumps are assumed as known constants. The heat absorbed in heating the working fluid to the saturated state is a small portion of the total energy taken up in the boiler; therefore, the sensible heating of the fluid is neglected.

1. Its Eciler

The triler consists of a number of tube-in-shell heat exchangers connected in parallel as shown in Fig. 2. Figure 4 shows the internal arrangement of one shell. The tute bundle is horizontal and boiling occurs on the outside surface of the tubes. The tube bundle is submerged in the working fluid. The working fluid is maintained at a level to just cover the top row of tubes. The warm seawater is jumped through the inside of the tubes.

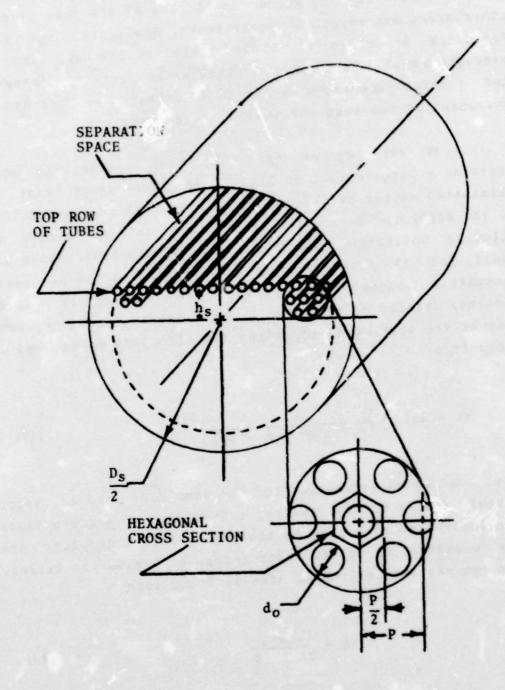


FIGURE 4. BOILER CROSS SECTION

The clearance between the tubes is fixed at 0.5 inches (1.27 cm) to allow for flaring of the tube ends or other space and strength requirements. Similarly, the tube thickness is fixed at 0.028 inches (0.071 cm), since no strength requirements due to differences between internal and external pressures on the tubes or material loss due to corrosion of the tube surfaces are accounted for.

At the tcp of the tube bank the leaving vapor contains a large amount of liquid droplets which must be eliminated either in the boiler or by an external separator. In the model the separation takes place in the boiler by allowing sufficient space above the tube bank. Palen and Small, Ref. 11, presented a method for finding the necessary separation volume that considers the effects of surface tension, density of the vapor and the liquid, and the mass flow of the working fluid. First the allowable vapor load is found from

$$VL = 2290 P_4 \left[\frac{\sigma}{6.86 \times 10^{-5} (\rho_4 - \rho_9)} \right]^{0.5}$$
 (12)

Valor load is an expression for the mass flow rate of vapor through a unit volume. The expression for the maximum vapor load limits the flow rate so that the moisture droplets are able to settle out of the vapor before it leaves the boiler. Then the required segmental area is found from

$$SA = \frac{M_{wf}}{VL L S}$$
 (13)

The segmental area is the cross sectional area from the top of the tute bank to the top of the shell. Knowing the segmental area the height of the tube bank above the main

diameter is found from

$$SA = c.25 \text{ Ls}^2 ARCOS(\frac{2 \text{ hs}}{Ds}) - \text{hs}(0.25 \text{ Ls}^2 - \text{hs}^2)^{0.5}$$
 (14)

Palen and Small also present a much simpler method of determining the total height of the tube bank. The height of the separation volume is set at 40% of shell diameter which results in "hs" being equal to 0.1 Ds. Either method can be used alone or the first method can be used as a constraint or the second. For this problem the segmental area is found by both methods with the first used as a check on the values given by the second.

Since the heat flux calculation is based on a single tule, the number of tubes per shell must be determined in order to find the total energy absorbed by the heat exchanger. The tubes are assumed to be laid out in a 60 degree triangular pattern for maximum compactness. The tules are located at the vertices of an equilateral triangle. Two cross sectional areas are calculated. The area of a lexagon whose minor radius is equal to one half the pitch is determined from

$$A = 6\left(\frac{P}{2}\right)^2 \text{ TAN 30}^{\circ}$$
 (15)

Then the effective cross sectional area of the tube bank is found by subtracting the clearance area, SA, from the cross sectional area of the shell after correcting the shell diameter for a clearance space to allow for free flow channels both within the tube bundle and around the ortside of the tube bundle. The effective cross sectional area of the tube bank is divided by the area of the hexagon, found

above, to determine the number of tubes in a shell.

The flow of the liquid and vapor through the tube bank is assumed to be turbulent. The incoming subcocled fluid from the feed working pump rixes with the recirculating fluid returning from higher up in the tube The circulation of the fluid is induced by the boiling process. This mixture enters the tube hank and begins the boiling process. Near the bottom of the tube bank the leat transfer goes mostly to bring the fluid up to the boiling temperature. After the fluid reaches saturated liquid conditions, stable boiling begins. The vapor bubbles through the surrounding liquid increase the turbulence around the tubes. As the liquid continues up through the tube bank, the local boiling temperature is decreasing due to the decreasing hydrostatic head. the boiling temperature increases the terrerature difference and increases the heat flow and increases the bubble generation rate. The vapor generation may become so rapid that a condition called vapor blanketing may cccrr in the interior of the tube bank where the liquid cannot flow inward fast enough to displace the wapor flowing region. This is similar to burnout in the convective boiling in that the tubes dry out and the heat transfer rate decreases. This effect becomes proncunced only near the critical heat flux according to Starczewski, in Ref. E.

All of the heat and mass flow variations in the tube bundle are averaged by performing the calculations for a single tube located at the mid height of the tube bank. This tube is assumed to represent the average of all the tubes in the tube bank. The boiling situation is assumed to be saturated pool boiling. Quoting from Collier, Ref 12, "pool boiling is defined as boiling from a heated surface submerged in a large volume of stagnant liquid". Since the

THE RESERVE THE PROPERTY OF TH

fluid volume seen by the tube is neither large nor stagnant, a correction is needed for the effect of being in a tube bundle. Saturated boiling is defined as boiling at a constant temperature and pressure as the liquid changes to a varor. This is probably not strictly true because the boiling temperature is controlled by the local pressure at each tute, and the fluid is not reaching temperature as it rises in the bundle. The situation is protably closer to convective boiling since the fluid is boiling in a confined channel between the tubes. additional effect in the boiler is the fact that the hydrostatic lead at the bottom of the tute bundle causes the toiling to take place at a higher pressure at the lottom than the tailing at the top of the bundle where the pressure is equal to the boiler exit pressure. This effect is accounted for by using the average head of the bundle in the heat transfer equations.

The heat transfer analysis of the boiler begins with the basic equation

$$Q = U \lambda \Delta T_{m} \tag{16}$$

where U is the overall heat transfer coefficient, it is the area through which the heat is transferred and ΔT_m is a properly defined mean temperature difference. In this paper, the product UA is treated as a single quantity called the thermal conductance. One may look at the equation as an analogue of the basis electrical equation I = E/R where I is equivalent to Q, E is the same as ΔT_m and UA is 1/R. This analogy is used when developing the expression for UA.

In crder to account for the effects of the tube seeing many nearby neighbors, the equation is modified by the inclusion of a factor, F, to account for the effects of

the surrounding tubes. Equation 16 is changed to

$$Q = UA F \Delta T_{m}$$
 (17)

where F is defined as the ratio of the heat flux of the tube burdle to the heat flux for a single tube in a stagnant pool booling situation. According to Ref. 13, a report published by workers at Heat Transfer Research, Inc. (HIRI) of Alhambra, California, F is greater than one. In Figure 5, which is sketched from a figure in their report, F ranges from 4.73 at $\Delta T_m = 8^{\circ}F$ to 2.63 at $\Delta T_m = 40^{\circ}F$. Since no data is given on the conditions of the test, F is assumed to be fixed at three.

In order to find ΔT_m , the heat exchanger effectiveness, ξ , is determined. The effectiveness is defined as the ratio of the actual heat transfer rate to the maximum possible heat transfer rate as shown by

$$=\frac{C_h \left(T_{h,in} - T_{h,out}\right)}{C_{min} \left(T_{h,in} - T_{c,in}\right)}$$
(18)

where

$$C = (M c_p)_{fluid}$$
 (19)

$$C_{min} = minimum of C_h or C_c$$
 (20)

In the boiler, $T_{h,in}$ is the temperature of the entering seawater, and $T_{c,in}$ is the saturation temperature of the working fluid at the pressure at the mid height of the boiler. Reference 14, by Kays and Iondon, contains equations for finding the effectiveness for various flow configurations. In addition to finding C_{min} , C_{max} must be

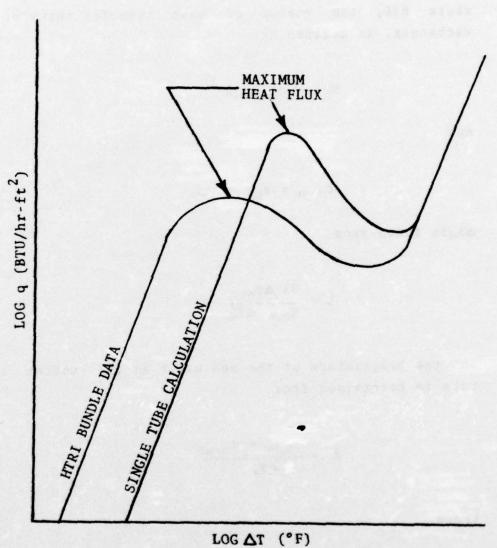


FIGURE 5. COMPARISON OF A SINGLE TUBE

VERSUS A TUBE IN A BUNDLE

known in order to decide which equation applies. For the case of boiling, C_{max} is infinite; therefore, regardless of the flow pattern

$$\xi = 1 - EXP(-NTU) \tag{21}$$

where NII, the number of heat transfer units of a heat exchanger, is defined as

$$NTO = UA/Cmin$$
 (22)

and

AImis found from

$$\hat{E} = \frac{UA \Delta T_m}{C_{min} \Delta T_o} \tag{24}$$

and the temperature of the sea water at the outlet of the tute is determined from

$$\xi = \frac{T_{hin} - T_{hout}}{\Delta T_{o}}$$
 (25)

where

$$\Delta T_o = T_{h,in} - T_{c,in}$$
 (26)

The quantity UA is determined next. Figure 6

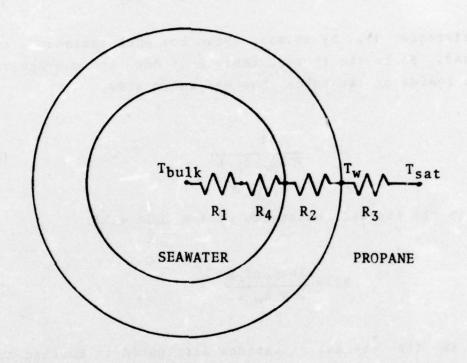


FIGURE 6. THERMAL NETWORK FOR A SINGLE TUBE

illustrates the method used. If UA is pictured as the reciprocal of resistance then the flow of energy from one fluid to the other can be pictured as a series electrical circuit and UA is determined from

$$UA = 1/(R1 + R2 + R3 + R4)$$
 (27)

Reference 15, by Holman, shows how each resistance can be found. R1 is the thermal resistance due to convection on the inside of the tubes, the sea water side.

$$R1 = \frac{1}{\pi h_{sw} d_j L}$$
 (28)

R2 is the thermal resistance of the tube wall.

$$R2 = \frac{\ln (d_o/d_i)}{2 \pi k_w L}$$
 (29)

R3 is the thermal resistance attributed to boiling on the cutside of the tubes.

$$R3 = \frac{1}{\pi h_{wf} d_o L}$$
 (30)

R4 is the thermal resistance due to any other resistances, such as fculing and corrosion deposits.

The heat transfer coefficient on the seawater side, h_{SW} , is determined from the dimensionless group called the Nusselt number defined as follows for flow inside of tubes.

$$Nu = \frac{h_{sw} d_i}{k_{sw}}$$
 (31)

In order to find the Nusselt number, three distinct regions of flow are allowed for in the problem: lawinar, transition, and turbulent flow. The division points are in terms of values of the Reynolds number, $Re = ud\rho/\mu$,

laminar: 0 < Re ≤ 2,000 transition: 2,000 < Re ≤ 10,000 turbulent: 10,000 < Re ≤ infinity

In the laminar region the Sieder-Tate correlation is used.

$$Nu_L = 1.86 (Re Pr d_i/L)^{1/3}$$
 (32)

In the turbulent region the Littus-Boelter correlation is used.

$$Nu_{T_4} = 0.023 \text{ Re}^{0.8} \text{ Pr}$$
 (33)

In the transition region the flow is unstable, but from data in Ref. 14 it appears that a fairly smooth curve drawn from the laminar curve to the turbulent curve adaquately represents the value of the Nusselt number in the transition region. This is done by assuming the form

$$Nu_{Tr} = y Nu_{L} + (1 - y) Nu_{Tu}$$
 (34)

where

$$y = a Re + b Re + c Re + d$$
 (35)

The coefficients in the polynomial are found by requiring

$$Nu_{Tr} = Nu_{L}$$
 at Re = 2,000
 $Nu_{Tr} = Nu_{Tu}$ at Re = 10,000

and that the first derivatives with respect to the Reynolds number match at Re equal to 2,000 and 10,000.

The I.I. Mostinski correlation, from Ref. 16, for single horizontal tube pool boiling is used to find the heat transfer coefficient on the outside of the tube. HTRI's extensive experimentation led Palen, Yarden, and Taborek, in Ref. 13, to conclude that Mostinski's equation produced the "most consistent results". His correlation is

$$h_{wf} = 0.00658 p_c^{0.69} q^{0.7} (1.8 r^{0.17} + 4 r^{1.2} + 10 r^{10}) (36)$$

where

r = F/Pe

 p_c = critical pressure of the working fluid (psia) q = heat flux (BTU/hr-ft²)

A final heat flux relation needed is the relation for the maximum heat flux allowed. Mostinski,s correlation for a single ture is used.

$$q_{\text{max}} = 803 \text{ p}_{\text{c}} \text{ r}^{0.35} (1 - \text{r})^{0.9}$$
 (37)

where the units are the same as in equation (36). From HTFI's data, under unknown conditions, q_{max} for the tundle appears to be one third of the q_{max} for a single tube.

The power required to pump the salt water through the tubes depends on the pressure drop across the tube bank. There are three causes of the pressure drop. The flow experiences a loss as the water enters into and exits from the tubes in addition to the frictional loss along the tube. Fice Ref. 14, the pressure drop for a liquid is determined from

$$\Delta_{\rm F} = 0.5 \, \rho_{\rm sw} \, u_{\rm sw}^2 \, (\frac{4 \, {\rm f \, L}}{{\rm d}_{\rm j}} + {\rm Kc} - {\rm Ke})$$
 (38)

In crder to determine f, Ke and Kc, the flow regime must be specified. The regimes are a little different than those for the Nusselt rumber.

laminar: 0 < Re ≤ 2,000 transition: 2,000 < Re ≤ 5,000 turbulent: 5,000 < Re ≤ infinity

The Fanning friction factor, f, in the laminar regime is

$$f = 16/Re$$
 (39)

In the turbulent regime, the Fanning friction factor is found from the Blasies equation in the range 5,000 < Re < 100,000

$$f = 0.079/Re^{0.25}$$
 (40)

which is a close approximation to the Karman - Niktradse equation,

$$(4 f)^{-0.5} = -0.8 + 0.87 \ln (Re(4 f)^{0.5})$$
 (41)

which is valid throughout the turbulent region (Re > 5,000), assuming smooth valled tubes. Either equation (40) or (41) may be used with similar results but the Flasius is better suited to this analysis because the Karman-Nikuradse equation must be solved by iteration. Equation (41) consumes more computer time to solve and can cause accuracy problems if not enough significant digits are evaluated. The friction factor in the transition region is determined by fitting a least squares curve through a hand drawn curve from the end of the laminar curve to the beginning of the turbulent curve.

The entrance effect, Kc, and the exit effect, Ke, depend on the Reynclds number, the free flow to frontal area ratio, ϕ , and, in the laminar region, on the tube length. For the kciler, ϕ is determined by

$$\phi = \frac{d_i^2 N}{DS - 4 SA} \tag{42}$$

Figure 9, page 7-13, in Ref. 17 is used to find Ke and Kc. The effect of Reynolds number and L/d ratio was eliminated from the relationships in the laminar and the turbulent regimes of the flow, but the Reynolds number was included in the transition region because of the large variation of Kc with Reynolds number. The following equations are used to find the entrance effect in the various flow regimes.

Laminar: (using the curve for 4 (L/d) Re = 0.1)

$$Kc_{\perp} = 1. - 0.4 \, \phi$$
 (43)

PROLESION.

Turbulent: (using the curve for Re = 5,000)

$$KC_{Tu} = 0.52 - 0.4 \phi$$
 (44)

Transiticr:

$$Kc_{Tr} = \lambda Kc_{L} + (1 - \lambda) Kc_{Tu}$$
 (45)

where

$$\lambda = e Re + f \tag{46}$$

The coefficients e and f are determined by requiring the transition Kc to equal the laminar Kc at Re equal to 2,000 and the turbulent Kc at Re equal to 5,000. The exit effect is the same for all flow regimes, and is found from the following equation that approximates the 4(L/d) Re = 0.05 curve in Ref. 17.

$$Re = 1.0093 - 2.5178 \phi + 1.1613 \phi^2 - 0.17677 \phi^3$$
 (47)

2. The Condenser

The design of the condenser is almost the same as the design of the boiler. It is a fixed tube sheet, tube-in-shell heat exchanger. The condenser does not need a separation volume so the entire shell is filled with tubes. In order to allow the entering vapor to reach all of the tubes, channels through the tube bank are needed. The allowance for this is accomplished by decreasing the shell diameter by a fixed factor when calculating the number of tutes. The calculation of the working fluid side heat transfer coefficient, h, is different. The equation accounts for the condensate from one tube falling on those below it and for the subcooling of the condensate. M.M. Chen's equation, as developed in Ref. 18 and shown in a more simple form in Ref. 17, is used.

$$h_{wf} = 0.728 \left[1+0.2 \left(\frac{c_p \Delta T}{H_{fg}} \right) (n-1) \right] \left[\frac{g \rho_t (\rho_t - \rho_g) k^3 H_{fg}^2}{n d_o \mu \Delta T} \right]^{0.25}$$
(48)

where

$$\Delta I = I_{Sat} - I_{w} \tag{45}$$

$$H_{fg}^{*} = H_{fg} + \frac{3}{8} c_{p} \Delta T \qquad (50)$$

n = number of tubes in a column

H_{fg} is the enthalpy corrected for subcooling of the cordensate. Equation (48) replaces equation (36) in the analysis of the condenser. In the circular shaped tube tark, n is found by determining the number of tubes in a column of average height.

$$n = 9.425 \frac{Ds}{pi} - 1$$
(51)

The definition of ΔT_o becomes

ed to lik domen or hoday polician but wells of manna and the contract the same and the same and

$$\Delta T_o = T_{sat} - T_{sw,in}$$
 (52)

in contrast to equation (26) above. The seawater cutlet temperature is found from

$$E = \frac{T_{sw,out} - T_{sw,in}}{\Delta T_o}$$
 (53)

which replaces equation (25).

In cider to find the heat transfer coefficient, the temperature of the outside tube surface is needed in equation (49); therefore, the thermal network is solved for the outside hall temperature and for the heat flow rate.

$$Q = Uh' (T_w - T_{bulk})$$
 (54)

where

$$UA' = 1/(R1 + R2 + R4)$$
 (55)

Toulk = average of the inlet and the outlet seawater temp.

There is no factor, P, in the heat transfer equation for condensation since the effect of neighboring tubes is accounted for in equation (47).

3. Ite Fusps

Iters are three pumps in the model. Poth circulation pumps for the boiler and the condenser are assumed to be propeller pumps (axial flow pumps). The feed

rumps are assumed to be centrifugal pumps. The differences in the characteristics of the two types of rump show up only in the calculation of capital costs of the rumps in the model.

The choice of the type of pump to apply is made on the basis of the flow rate, GPM, and on the head the pump is working against. Qualitatively, the propeller pump is used where the flow rates are high and the head is low. The centrifugal pumps apply where the flow rate is comparatively low and the head pressure is high. The qualitative picture is placed in better perspective through the calculation of the specific speed from

$$Nsg = \frac{\text{BPM GFM}}{0.75}$$
Head

The centrifugal pump is used when $500 < Nsq \le 7,500$ and the propeller pump applies when $7,500 < Nsq \le 15,000$, according to Ref. 19. The choice of the type of pump is checked after the optimization process has been completed by solving equation (55). If the value of the specific speed is outside of the range, 500 < Nsq < 15,000, the maximum efficiency of the pump drops off rapidly; therefore, few rumps are designed to operate outside of this range.

Since the pumps are "black boxes", only the power consumed by the pumps is calculated. The power required to drive the pump is found from

$$P = \frac{H \Delta p}{\gamma_{pump}} \tag{57}$$

issuabang aty the tallog and jot ages; asin the care

D. COST FFAMEWORK

As discussed at the start of this report, the capital costs of the major components are the only costs to be considered in this model. The boiler and the condenser are both tute-in-shell heat exchangers assumed to be of the fixed tute sheet type. The only difference being that the shell of the boiler is partly filled with tubes. Both seawater circulation rumps are assumed to be propeller type rumps and made of the same material. The size and number per heat exchanger shell may be different for the boiler and the condenser. The number of working fluid feed rumps, which are the centrifugal type pumps, are assumed to equal the number of boiler shells.

The cost estimating relationships for the heat exchangers and the centrifugal rumps are taken from Ref. 20, by K.M. Guthrie. The equation for the cost of the heat exchangers is

$$$he = 118.3 (F_d + F_p) F_m I_{he} (A_t)^{0.63}$$
 (58)

where the constants are calculated from a log-log plot of cost versus total tube surface area and

Fd = 0.8 for a fixed tube sheet

Pp = 0.0 for pressure correction (p < 150 psi)

Pm = material factor from Table 1

Ine = ccst index

At = do L N

for a relevant range of 100 < A < 10,000 ft2.

The cost equation for the centrifugal pumps and the

associated motors is

$$f_p = 1.013 P_m P_0 I_{fp} C_H^{0.721}$$
 (59)

where the constants are found from a straight line approximation of the log-log plot of cost versus C for the section of the curve from C = 30,000 to C = 300,000 for an electric motor driven pump and

Fo = 1.0 fcr suction temperatures < 250°F (121°C)

Ifp = ccst index

Fm = waterial factor from Table 2

C_H = product of the flow rate in GPM and the pressure differential in psi

for a relevant range of 30,000 $< C_H < 300,000$. The cost data in Guthrie's article is given for a time base of mid-1968.

Reference 21, also by Guthrie, gives cost data for propeller rumps and centrifugal pumps. The equation for the propeller rumps with the motors is

where the constants are found from the log-log plot of cost versus flow rate for an electric motor driven pump and

Fp = 1.0 fcr a suction pressure < 150 psi

Fm = material factor from Table 3

Ich = cost index

Flow = flow rate in GPM

for a relevant range of 1,000 < Flow < 100,000 GFM. The time base for the propeller pump data is for the erd of

Table 1
MATERIAL FACTORS FOR THE HEAT EXCHANGERS

Material

Shell	<u>Tube</u>	<u>F</u> m
Carton Steel	Carbon Steel	1.00
Carton Steel	Brass	1.52
Carton Steel	Stainless Steel	3.52
Stainless Steel	Stainless Steel	4.50
Carton Steel	Monel	3.75
Mcnel	Monel	4.95
Carbon Steel	Titanium	11.10
Titarium	Titanium	16.60

Table 2
MATERIAL FACTORS FOR THE CENTRIFUGAL FEED PUMPS

Material	<u>Fm</u>
Cast Iron	1.00
Bronze	1.28
Cast Steel	1.32
Stainless Steel	1.93
Monel	3.23
Hastelloy C	2.89
Titanium	8.98

Table 3 MATERIAL FACTORS FOR PROPELLER PUMPS pumps

Material	Εm
Cast Iron	1.00
Cast Steel	1.28
Stainless Steel	1.64

1970. The costs arrived at from equations (58), (59), and (60) are in dollars per component, for example, dollars per heat exchanger shell. The purpose of the cost index is allow all the equipment to be costed on a common basis. The time difference is assumed to be small enough so that the cost index is set to 1.00 for all the components.

E. ASSEMBLY OF THE MATHEMATICAL MODEL

The engineering and the cost relationships are linked together in the model. The cost equation is the function to be minimized and the implicit and explicit constraints are the engineering equations that restrict the feasible region of the problem. The explicit constraints restrict the variables, the vector X, to positive values. The rest of the model is formulated in the following form

subject to:
$$g_i(X) \ge 0$$
, $i=1,2,...,m$

$$h_i(X) = 0$$
, $i=m+1,...,m+n$

where the definitions of the elements of X are shown in Table 4. Several of the variables must have only integer values if the system were actually built, but for the objectives of this analysis they are left as continuous real variables. For example, the number of shells may have an integer value in the actual system but this requirement can be accomplated by solving the system for a real value then integerizing the number of shells and solve the problem with the next higher and the next lower integer number of shells. The lower of the two solutions would be chosen. The final resolution of the correct number of shells would

Table 4 DEFINITION OF THE ELEMENTS OF THE VECTOR X

- x(1) = velocity of seawater in the boiler
- x(2) = " " condenser
- x(3) = cutlet pressure in the boiler
- x(4) = inlet " " condenser
- x(5) = tube length in the boiler
- x(6) = " " " condenser
- x(7) = mass flow rate of the working fluid
- x(8) = diameter of the boiler shell
- x(9) = " " condenser shell
- x(1C) = outside dia. of a tube in the boiler
- x(11) = " " " " condenser
- x(12) = number of boiler shells
- x(13) = " condenser shells

reflect to lagrant to book postibles on the total set

probably depend on other than thermodynamic and cost considerations, such as a requirement that one extra shell be included to allow for one shell being down for maintenance. Obviously, the number of tubes in a shell must be an integer value but one extra tube cut of several thousand is a minor matter. In addition, there may be a requirement for extra tubes to allow for a percentage to be plugged when they develop leaks so that the entire shell is not put cut of commission by a minor leak in one tube.

1. Chjective Function and the Constraints

The chjective function is the sum of the costs of the boiler and the condenser, and the pumps. The model assumes that there is one circulating pump and one feed pump for each shell in the boiler and that there is one circulating pump for each shell in the condenser. Since there is no cost analysis for the turbine, no assumption is needed about the number of turbines in the system.

There are nineteen constraints in the present program, one of which is optional. The first three concern the saturation pressures in the boiler and the condenser. The next twelve place restrictions on various dimensions of the toiler and the condenser. The last three require that some intermediate calculations stay positive.

Ising the objective function as a constraint has proved advantageous in cases where the problem was non-convex or where the program had difficulty in finding a solution. It is used by setting a boundary on the objective function so that the constraint is either infeasible or slightly feasible at the starting point. The proper choice must be found by trial and error. One caution to note is when equality constraints are involved. If the starting

from the boundary and the equality constraints may not be satisfied when the program finishes.

The need for the other eighteen inequality constraints is not obvious when the problem is first formulated. The thermodynamics of the problem require, for example, that the propane saturation temperature he less than seawater temperature in order for the boiler to absorb energy, but the program has no way of knowing that, unless it it specified. If the program can improve the cost and satisfy the other constraints by reversing the relationship it will do so. This inversion may take place only for a short time during the solution of the problem, but, if the equations fail under this condition, the constraint on the temperatures in the boiler is necessary. Most of the other constraints are needed for the same reason.

Some constraints are not added to the problem since they are not binding in the solution. There are two such constraints. The first is the segmental area required to allow separation of the liquid droplets from the vapor. The segmental area set aside by the 40% rule-of-thumb is more than that required by the consideration of the physical properties of propane. The other constraint is the restriction on the maximum heat flux by Mostinski's correlation, equation (37). The constraints do not become binding in the solution.

The three equality constraints contain the calculations of the properties of the various components modeled in this analysis. The first one concerns the boiler. It states that the actual energy absorption rate of the boiler shells must be equal to the energy absorption rate required by the thermodynamic cycle. Similarly, the second equality constraint states that the heat rejected by

the condenser must equal that required by the thermodynamic cycle for the condensation of the vapor from state point 5 back to state point 1 as shown in Fig. 3. The third constraint calculates the power required by the various rumps and the power extracted from the turbine to run the system's rumps and to produce electrical power for consumption. This constraint requires that the power cut of the turbine equal the power used by the rumps and the generator. The net useful power cut of the system is a design objective and is specified as 25MW herein.

The Cptimization Method

Since the arrival of the electronic computer, a great many nonlinear programming techniques have been developed from early theories and more current research. The computer program, called the Sequential Unconstrained Minimization Technique (SUMT), developed by Mylander, Holens, and McCormick in Ref. 22 is used to optimize the Their program implemented much of the theory contained in Ref. 23, by Fiacco and McCormick, about nculinear programming using unconstrained minimization techniques. SUMT-Version 4 is chosen because of its atility to solve a wide variety of problems. The theoretical requirements for SUMT to be guaranteed to find a local minimum is that there exists some point which satisfies the inequality constraints and that there not be a local minimum at prints where X goes to infinity. If certain convexity conditions are satisfied, then the local minimum found is a glebal mirimum.

SUMT uses a penalty function to transform the constraired optimization problem

subject to:
$$g(x) \ge 0$$
, $i=1,2,...,m$

$$h(x) = 0$$
, $i=m+1,...,m+n$

into a sequence of unconstrained problems of the form

$$F(X,r) = f(X) - r \sum_{i=1}^{m} ln g_i(X) + 1/r \sum_{i=m+1}^{m+n} (h_i(X))^2$$
 (61)

where F(X,r) becomes the function to be minimized. inequality constraints, g(X), cause a large renalty to occur as a boundary is approached. The equality constraints add a large recalty whenever h(X) departs from zero in either direction. The size of the renalties due to g(X) and h(X)is controlled by the arbitrary parameter "r". The algorithm solves the problem by solving a series of subproblems. $r_1 > r_2 > r_3 ... > 0$, SUMI minimizes $P(X(r_n), r_n)$ in each subproblem in order to find the minimum of f(X) as "r" tends to zero. When a minimum of a subproblem is found, "r" is reduced by a factor assigned by the user and the new subproblem is sclved. The process continues until the convergence criteria chosen by the user is satisfied. illustrates schematically the results of the sequence of sulproblems and the qualitative shape of the penalty function. Three methods of minimization are included in the program. The user may choose either the generalized Newton-Raphson, steepest descent, or McCormick modification of the Pletcher-Powell method.

There are several other options that control the operation of SUMT. The user may choose the initial penalty, r, cr, if the problem satisfies certain conditions, the user may elect to have SUMT compute the initial "r". There are three choices for the convergence criterion of the

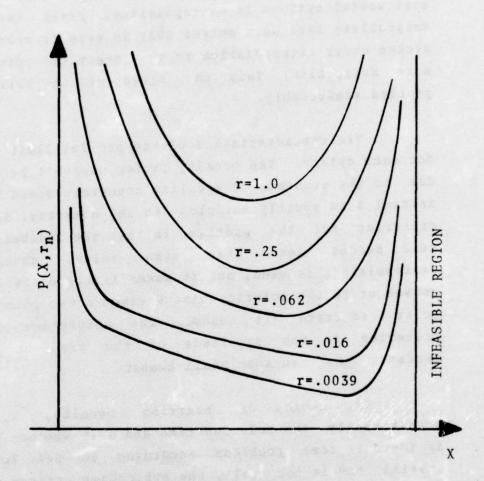


FIGURE 7. PROBLEM SOLUTION SEQUENCE

problem. The criterion used depends on the inequality penalty term being close enough to zero. Another option is the choice of the calculation of the subproblem stopping criterion. Two of the choices compute the inverse of the second partial matrix and the third choice computes only the gradient of the penalty function ($\nabla_X P(X,r)$). One of the most useful options is extrapolation. After two or three subproblems have been solved SUMT is able to make a first or second order extrapolation to the starting point for the next subproblem. This can speed up the solution of the problem measureably.

The characteristics of the problem limit the choices for each cpticn. The penalty factor couldn't be set by SUMT due to the presence of equality constraints and because the initial X is usually not close to any boundary. The stopping criterion for the problem is that the inequality penalty term become less than some value, usually 10-7. Extrapolation is used, but it makes little difference in the answer or to the solution time whether a two point or three point estimate is made. The subproblem convergence criterion is the magnitude of the gradient of F(X,r) becoming less than some small number.

The choice of starting penalty, r, affects significantly the solution time and even whether a solution is found in some problems according to Ref. 7. If the initial "r" is too small, the subproblem optimum may be the problem's optimum which may be hard to find or require more calculations than if a much larger "r" is chosen. If "r" is too large, the first few optima lie near the center of the feasible region and are unrelated to the optimum of the objective function.

Scaling of the variables, constraints, and the objective function is necessary. The variables must be

within two cr three orders of magnitude of each other. For example, the working fluid flow rate is of the order of 107 whereas the tube diameters are of the order 100. If the finite difference interval is 10-4, the partial of the equations with respect to the mass flow rate may be lost due to the roundoff error providing the only significant figures. With the constraints and the objective function, unscaling may result in one term in the penalty furction being dominant, and the algorithm finds that most of the decrease in the penalty function takes place in the dominate term. The result is that SUMT works hard to decrease that term and does not see the other terms resulting in a meaningless solution.

In making first and second order estimates of the sclution, SIMI may generate negative numbers in the X vector. When it evaluates the constraints and objective function for this point, the equations fail and may cause the program to fail depending on the severity of the failure and how many times the error has occured. This problem is circumvented by testing the vector upon entry into the subroutine where the objective function and the constraints are evaluated and setting the constraint being evaluated equal to some negative constant.

The main program of SUMT is altered so that the problem can be restarted at the current solution without reading in a new problem. This is necessary because the program would give all the indications of having found a solution, but if the problem is restared it could often find a low value for the objective function. The new solution is compared to the previous solution and if the two solutions are not close enough the problem is restarted after making some changes to particular parameters, options, or constraints depending on the objective the user has in mind.

III. RESULTS AND CONCLUSIONS

One of the major tasks in the development of a nonlinear model is the linking of the model with the optimization program in order to analyze the model. The selection of the proper nonlinear programming technique is essential to the solution. Several techniques require that the objective function and the constraints be in a particular form. For example, Geometric Programming developed by Euffin, Peterson and Zener, in Ref. 24, requires that the equations be posynomials which is not true of the model developed in this paper. SUMI was chosen because of its wide applications to problems of any type. It only requires that there exist some feasible point and that local minima not exist at infinity. It has successfully solved many nonconvex problems.

Two versions of the analytical model are presented. called the iterative version because the calculation of the heat absorption rate and heat rejection rate require iterating on some quantity to find them. Mostinski's correlation, equation (36), contains the heat flux which is an unknown. It is not possible to solve the system of equations explicitly for the heat flux; therefore, successive approximation technique is used. calculation of the heat rejected in the condenser involves the same process with the temperature of the outside tube wall as the iterating variable. SUMT is unable to arrive at a consistent solution in the time allowed, up to 90 minutes of computer time. The solutions are quite different for each try with nc pattern evident about the direction the sclution might lie.

are several possible explanations for this behavior. Ferhaps the behavior of this version of the problem is such that SUMT is not the best choice for the optimizing algorithm or possibly the problem cannot be sclved at all in this form. Another source of problems could be the numerical differentiation procedure. differentiation subject to severe truncation and is round-off error. The subproblem always terminates when SUMT is unable to reduce the penalty function during the next iteration, from one estimate of X to the next estimate. Remember the convergence criterion, never satisfied for this version of the problem, is the magnitude of ₹(X,r) becoming less than some small number. No conclusive results could be obtained with this version so the second version is used for the present analysis.

In this second version the heat transfer coefficients on the Frogene side of the tube are fixed. This enables the equality constraints for the boiler and the condenser to be calculated to the full accuracy of the computer without increasing the time required. The correlations, equation (36) and (48), are used for comparison with the fixed value. If the calculated value and the fixed value are within 20 to 25% of one another, the solution would be considered to be good since this is the tolerance claimed for the correlations.

SUMT seemed to be able to solve the new problem because a series of seven solutions have been produced where the values of the elements of X and of the objective function vary no more than 3.7% and 1.2%, respectively, about the mean when different starting points and different SUMT control parameters are selected. See Table 6 for a summary of the solutions obtained from the assumed data, as shown in Table 5. The design implications of these results are quite reasonable. The range shows the spread in the solutions.

Table 5 PROBLEM INPUT DATA

Fouling Thermal Resistance	0.0 hr-of/BTU
Eciler and Condenser	(0.0 °C/kW)
Elevation of Boiler	15 ft
Above Condenser	(4.57 m)
Channel Allowance	0.83 ft
	(0.25 m)
Efficiencies for the Turbine	85%
and the Pumps	
Seawater Temperatures - Hot	75°F
	(23.9°C)
- Cold	40°F
	(4.4°C)
Net Output Power	8.532x107 BTU/hr
	(25 MW)
Fundle Factor (F)	3.0
Cost Indexes (I)	1.00 for all comp.
Tute Wall Thickness	0.028 in
	(0.071 cm)
Mir. Listance Between Tubes	0.5 in
	(1.27 cm)
Heat Transfer Coefficients	145 BTU/ft2-hr-0F
fcr Eciler and Condenser	(0.822 kW/m2-oc)

Materials

STANCE.

Eciler and Condenser	Carbon Steel Shells
	Brass Tutes
Feed Pumps	Bronze
Circulation Fumps	Stainless Steel

100 TANA 10

Table 6
EXAMPLE PROBLEM SOLUTION SUMMARY

Variagle	Range	Mean	Var	<u>Units</u>
			3	
Bciler:				
Velocity	3.323-3.347	3.335	0.7	ft/sec
	(1.013-1.020)	(1.016)		(m/sec)
Fressure	affrox 0	118.2	0.0	lb/in2
		(815.0)		(kp)
Tuke length	39.16-40.00	39.71	2.1	ft
	(11.91-12.19)	(12.10)		(m)
Shell dia.	24.12-25.00	24.91	3.5	ft
	(7.352-7.620)	(7.593)		(m)
Tube dia.	0.756-0.761	0.758	0.7	in
	(1.919-1.934)	(1.926)		(CA)
* of shells	7.141-7.409	7.228	3.7	
Cordenser:				
Velocity	2.793-2.823	2.808	1.1	ft/sec
August 2	(0.851-0.860)	(0.856)		(m/sec)
Fressure	91.53-91.60	91.56	0.1	lb/in2
	(631.1-631.6)	(631.3)		(kp)
Tube length	39.35-40.00	39.78	1.6	ft
	(11.99-12.19)	(12.12)		(m)
Shell dia.	44.69-45.00	44.89	0.7	ft
	(13.62-13.72)	(13.68)		(m)
lub∈ dia.	0.7632-0.7686	0.7660	0.7	in
	(1.938-1.952)	(1.946)		(cm)
of shells	3.297-3.401	3.331	3.1	
Mass flow rate	22.12-22.21	22.17	0.4	10 01tm/hr
	(10.03-10.07)	(10.06)		(10 %kg/hr)
System cost	7.578-7.673	7.607	1.2	106 \$

The arithmetic mean of the seven solutions is calculated. The difference between the maximum and the minimum values is shown as a percent of the mean. The percent variation gives some idea of how well the solution was determined. Note that the tute length and shell diameter constraints are binding for both the boiler and the condenser. The subgroblem still terminates only when the penalty function does not decrease from one iteration to the next. The magnitude of $\nabla_{\!\!X} P(X,r)$ is not becoming small enough to satisfy the convergence criterion. The value never becomes less than 0.02 and is usually greater than 0.5 when a subgroblem is stopped by the program.

Useful results are obtainable, however. The encineer can learn much about his problem even from a single point. For example, from the cost breakdown in Figure 8, the portion of the costs attributable to each component can be seen. The heat exchangers, of course, dominate with 78% of the cost. The circulation pumps are about 21% of the cost with the feed pumps being 1.2% of the total cost. Since the efficiency assumed for the circulation pumps is probably too high, these pumps would be even more costly in terms of energy and money. This could indicate the priorities that should he set on the design efforts for improvements of the various components with the highest priority being the highest percentage cost component.

The system's power distribution is shown in Table 7. The significant features are the large energy absorption rate for the small useful output power and the parasitic pumping power. The lower the efficiency the greater must be the power absorbed for the same output. This is to be expected for this kind of system. The parasitic pumping power (12% of the turbine output) is about twice what it is for a fossile fuel plant.

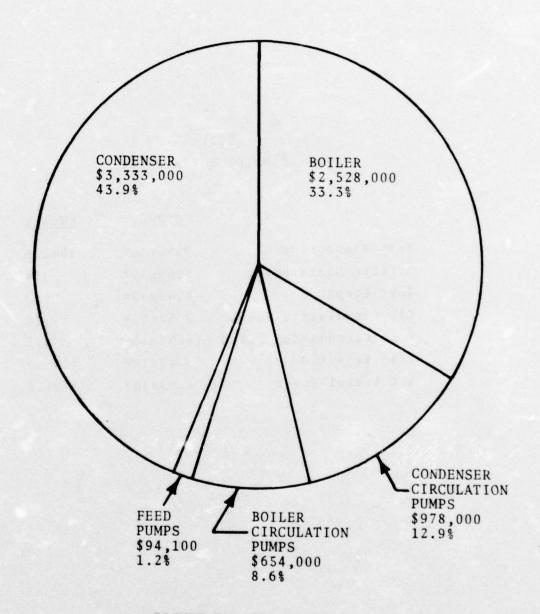


FIGURE 8. COST BREAKDOWN

Table 7
SOLUTION FOWER BUDGET

	BIUZDI	MWatts
Heat Absorption	3.56x109	1042.8
Turbine Shaft Power	9.69x107	28.4
Feed Pumps	4.49x106	1.3
Flr Circulation Pumps	2.58x106	0.8
Cond Circulation Pumps	4.51x106	1.3
Feat Fejected	3.47x109	1015.7
Net Useful Power	8.53x107	25.0

Cnce the model has been optimized with the number of shells as real variables, the problem can be re-solved twice by fixing the shells at the next higher and the next lower integer number. For the higher number of shells, the cost increased slightly as expected, from 7.607 million dollars average to 7.983 million dollars. The problem with the lower integer number of shells could not be solved due to the constraints on the tube length and the diameter of the shells being binding before the shell numbers were integerized. The program is unable to find a new solution that satisfies the equality constraints.

Table 8 shows some miscellaneous data from the mcdel comment. The overall heat ccefficient of the boiler is approximately three times that for the condenser which is the major reason for the lower cost of the boiler relative to the condenser. remember that the bundle factor, equal to 3, is a part of the heat transfer coefficient. More data must be gotten about this factor in order to improve the reliance or any There is a significant difference between the Propane heat transfer coefficients that were assumed and the theoretical values calculated from the correlations. Other values should be tried until the fixed value is within the 20% tolerance on the theoretical value. The flcw of seawater through the tubes in both the boiler and the ccrdenser is turbulent. Apparently, the resulting cost savings due to the isproved heat transfer rate through a smaller area is more than the increased costs of the higher pumping power needed. It seems likely that the solution obtained is globally optimal; several different starting prints were tried. The problem has been started in both the laminar and the turbulent flow regions with no change in the results.

Table 8
ADDITIONAL DATA FROM SAMFLE
PROBLEM SOLUTION

	Boiler	Condenser	<u>Units</u>
Working Fluid Heat Trans	sfer Cceffi	cients	
Assumed	145.	145.	BTU/hr-ft2-0F
	(0.822)	(0.822)	(kW/m2-cc)
Theoretical	227.	241.	
	(1.29)	(1.37)	
Surface Area per Shell	2.39x105	1.27x106	ft ²
	(2.22x104)	(1.18x105)	(m²)
Seawater Cutlet Temp	72.3	43.8	OF
	(22.4)	(6.55)	(°C)
Shell Saturation Temp	66.7	49.9	oF
	(19.3)	(9.94)	(°C)
Seawater Mass Flow Rate	4.39x108	9.11x109	lbm/hr
	(1.99x10e)	(4.13x109)	(kg/hr)
Mean Temp Difference	6.67	7.85	oF
	(3.71)	(4.36)	(°C)
Overall Heat Transfer	310.6	105.0	ETU/hr-ft2-0F
Coefficiert	(1.76)	(0.595)	(kW/m2-0C)
Heat Flux	2070.0	824.9	BTU/ft2-tr
	(6.528)	(2.602)	(kW/m²)
Seawater Reyrolds ro.	19,800.	9,960.	-

IV. RECOMMENDATIONS

The present model has a great deal more information to offer, but some improvements in the model should be made to improve the ability of a nonlinear program to solve the problem. The first recommendation would be to study the equality constraints and place inequality constraints on any variable or grouping of variables if they both can become negative and also must have their logarithms computed. Examples, already included in the program, are the number of tules in the shells. The heat transfer coefficients for the working fluid side in the boiler and the condenser could be included as equality constraints. This can be done by adding the heat transfer coefficients to the vector X and requiring that the variables be equal to the coefficients as computed by the appropriate correlation. The same could be done with Falen and Small's property-dependent calculation of the separation area by equation (12) and (13). cverall standgeint, the results seem to indicate either the problem is too difficult to be solved efficiently ir its present form or that another nonlinear program, other than SUMI, might be better suited to the problem. differentiation of the constraints in this problem do not seem to help the program to find the solution since the stcrs.

Once the optimization scheme is functioning properly, the model can be used to study the effects of fouling and corrosion and the costs of their prevention. Many proposals have been made that must be analyzed in the context of the total system. Some such as Americap, where plastic fram balls are pumped through the tubes to clean the heat

transfer surfaces, add a cost penalty from the capital and crerating cost of the Amertap system and from the increased pumping power required. Coatings add the capital costs of the coating and the added capital cost of increased heat transfer area required because of the increased thermal resistance of the coating.

A sensitivity analysis can be performed on the model to test the assumptions and the various constants. The sensitivity to changes in these factors can indicate how much one should be willing to pay for the improvement sought. High sensitivity indicates areas where the greatest effort for improvement should be desired or the priorities should be set on research into the unknown areas of the problem. The trade-off between cost and performance can be studied by changing the output power and looking for the lowest cost per kilowatt-hour, which is the measure by which conventional power plants are compared. One could look into the cost and the benefits of increasing the depth from which the cold water is pumped versus the improved temperature differential possible from using the colder water.

For the capital cost of the system to be complete, the turtine design must be included and cost estimates made. The turtine will be unique due to the very low pressure potential drcp it must use. Some work has been done by other groups working on OTEC, Ref. 25 and 26. Reference 27, by Robert I. Bartlett, can provide costing and design information although he deals only with steam turbines. The remaining part of the system that is desireable for inclusion in the analysis presented herein is the cold seawater intake pipe. It will be on the order of 1,000 ft (30.4.8m) long and 100 ft (30.4 m) in diameter.

```
FORMAT
              CARC
             X VERCTOR
35
```

U

```
CLER THE PREVIOUS SOLUTION. ALSO MOVE THE BOUNCARY OF THE DEJ FUNC
CONSTRAINT CLOSE TO THE FRESENT SOLUTION.
                                                                                                                                                                                                                                                                                                                             NTI,NT2,NT3,NT4,NT5,NT6,NT7,NT8,NT5,NT10
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                        CALL STERE
CALL STERE
IF (NEXCPI-GT-1) CALL CHCKER
IF (NEXCPI-EQ-3) STOP 01072
IF (NEXCPI-EQ-5) STOP 01104
CALL FEAS
CALL STERE
CALL STORY OF STORY EXIST
                                                                                                                                                                                                              N.M.MZ, TWMAX, RECIN, RATIC, EFSI, THETAO
                    CWS PROGRAMMER CENTROL
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                           (CABS($TOT-$Y).LE.$TCT/1.E05) GC TG

= $TCT/

= $TCT/1.E07+.0001

N = MN.H1
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                         MRITE (6 TO) AFR

STITE (6 TO) AFR

STITE (6 TO) AFR

STITIC COTA AFR

CALL COTA AFR

STIC COTA AFR

CALL COTA AFR

STITIC COTA AFR

CALL CO
                              TINE REACTIN IS UNDER FRUGER F
                                                                                                                                                                                                    C--FEACTION CO.
SLEBCLT
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                          NFFE
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                   40
                                                  UU
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                    U
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                         CCCCC
```

E 7 23 7

```
ZZZZZZZZZZZZZZZZZZZZZZ
                                                                                                                                 (1+0,5x,2HA=13,6x,2HP=13,6x,3+MZ=12/8x,1CHPAx, TIME=E14-7,MAI
(1+0,5x,5+RATIO=E14-7,6x,8+EPSILON=E14.7,4x,6HT+ETA=E14-7,MAI
                                                                                                                     SECOND SET OF CPTIONS OUTINE-SUMT VERSION 4
                                                                               (SE12-0, 314)
(SE12-6)
(SE12-6)
(SE12-6)
(SE20-7)
(1017)
(13+0 TCLERANCES )
(26+0 SECOND SET OF CPTICNS )
(S6+1 NONLINEAR PROGRAMMING ROUTINE-SCWT VERS)
                                        FORMAT
                                         CARC
                                  CARD FORMAT
                                                                                                                            150 FCFN AT 150 14x 2HR = 1 -
05
```

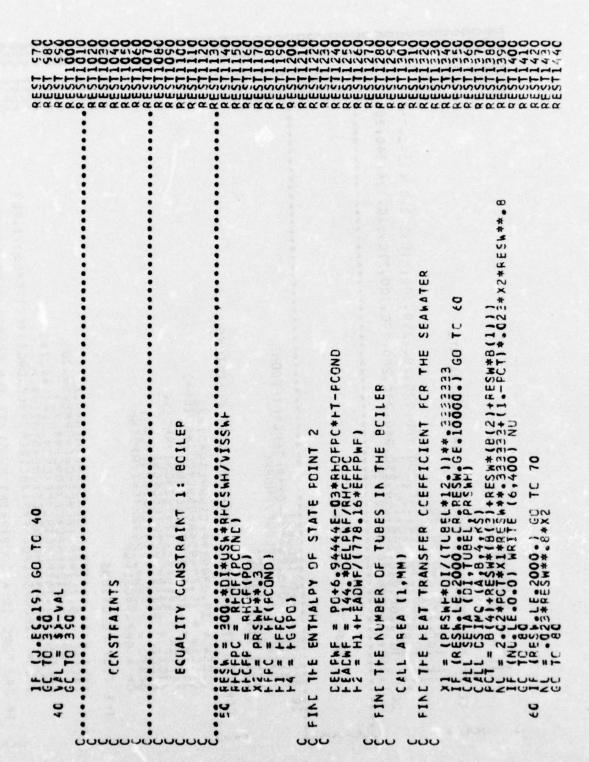
```
THE CBJECTIVE FUNCTION AND THE AS REQUESTED BY THE SUMT PROGRAM.
(JAVE)
    EVALLATES AT A TIME
RESTNT
    CCNSTRAINTS CNE
SLEFOLTINE
                                                                                           SET
                                                                                                     -
```

COCOCO

64*FMBLR*CEXBLR*ATOTB**.63 5.767*FMCIRC*CEXCRC*GPMSh**.783 1.013*FMFC*CEXCD*C+**.721 5.767*FMCIRC*CEXCRC*CGFMSh**.783 4.64*FMCOND*CEXCNC*ATOTC**.63 14.64*FMCOND*CEXCNC*ATOTC**.63 THE VARIABLES FOR THE COST FLACTIONS REQUESTED CRJ FUNC OR CCNSTRAINT 9.63495*RHOSMH*CSW*D1**2 19.63495*RHGSWC*CUSW*CIC**2 2618#OD*TUEEL*ENI CBJECTIVE FUNCTION CI = 0C-2.0*THICK CCST FLACTIONS d 13 AREA AFEA GFFF # 8 (18 E CEJ FUNC THE TOTOTOTO OTOTOTOTO SANDANA SANDANA SANDANA OTOTOTOTO SANDANA OTOTOTOTO OTOTOTOTO OTOTOTO OTOTO OTOTO OTOTOTO OTOTOTO OTOTO OTO CCPPLTE 9 21 OCCULUN -39 11 35 =

COO

UUU



```
DUTLET TEMP
                                                                                                                                                                                                                                                                                                                                ANE THE SEALATER
                                                                                                                                                                                                                                                                                                                                                                             = BFSWIN-DELTCB*(1.0-CEXP(-NTUB))
= UAB*FACT*CELTUB*(1.0-DEXP(-NTUB))/NTLB
E*CTCBE**7
(BTSWIN+TSMOB)/2.-T2
= QTUBE**S**BNI
                                                                                                                                                                                                                                                                                                                                                                                                                                                                   FATE
                                                                                                                                                                                                                                                                                                                                                                                                                                                                   THERMODYNAMIC CYCLE PEAT ABSCRPTICH
                                                                               72/(HCSN#DI*TUBEL)
55#DLOG(DC/DI)/(BKMET*TUBEL)
•63495*RFCSNH*USW*DI**2
N*CPSNH
S*5*DS
7222E-03*RFOFP*FEIGHT
                                                                                                                                                                           ShT = FLGSW*BNI$S

= CC*TUBEL

= 0 168131*PCRIT***65

= 10 6*R**17+4**10

= AP*RC/AA**7

T = 300

TCB = BTSWIN-TSAT(P)
                                                   CCMPLTE THE VARIOUS THERMAL RESISTANCES
                                                                                                                                                                                                                                                                                                                             FINE THE TOTAL HEAT TRANSFER RATE
7C NL = 1.86*RESM**,3323334X1
IF (NU.LE.2.66) NU = 3.66
EC FCSh = 12.*NU*CKSHF/DI
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                               GC TO 350
                                                                                                                                                                                                                                                                                        = 145.
= 3.81572/(HBA+AA)
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                        CALCLLATE THE CONSTRAINT
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                             CTCT = FLCNF* (H4-H2)
                                                                                                                                                                                                                                                                                                                                                          (AC+R3B)
                                                                                                                                                   7 PCR 1
                                                                                                                                                                                                                                                                                                                                                                                                                                                                   FINC THE
                                                                                                                                                                                                                                                                                                                    UUU
                                                                                                                                                                                                                                                                                                                                                                                                                                                       JUU
```

```
PRShC**3
PRShC**3
PRShC**3
E5W.LE.2000...CR.CRESW.GE.10000.) GC TC 100
ETA (C1D,CTUBL)
SIMC (A,B44,T)
SIMC (A,B4,T)
SIMC 
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                     F5S = F1+(ENTG(T2)-ENTF(T1))*(HGFC-HFPC)/(ENTG(T1)-ENTF(T1))
F5 = F4-EFFTUR*(H4-H5S)
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                           3.81972/(CHS%*CIC*CTUBL)
.159155*DLCG(COC/CID)/(CKMET*CTUBL)
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                 FINC THE SEAWATER HEAT TRANSFER COEFFICIENT
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                              FINC THE NUMBER OF TUBES IN THE CCNCENSER
      2: CONCENSER
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                            23*CRESW** 84CX2
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                          FINC THE ENTHALPY OF STATE POINT
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                               02*CRESW** 333333*CX1
LE-3*66) CNU=3*66
2**CNU*CKS%C/CIC
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                         FINE THE THERMAL RESISTANCES
      CONSTRAINT
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                CALL AREA (2,MM)
ECUAL ITY
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                               . .
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                               110
```

UUU

A CAS DESCRIPTION OF THE PARTY OF THE PARTY

UUU

```
THE SEAMATER OLTLET TEMP, AND THE
                                                                                        RHOFPC-RHCGPC1+TKF(T11++3
                                                                                                                                                                                                                                                                                                                          FINE THE THERMODYNAMIC HEAT TRANSFER RATE
HESMC+CUSH+CIE++2
                                                                                                                                                                                                                                                 I+CTACUT)/2.+CGTURE*BC
                                                                                                                                                                                                                                                                                             728*C*(BE*CHFG/CDELT)**.25
CTUBE*CS*CNI
                                                                                                                                                                             T TRANSFER RATE,
                                                                                                                                                                                                                                                                                                                                                                                 CC TO 350 TO 350
                                                                                                                                                                                                                     EXP (-CNTU))
                                                                                                                                                           .81572/1CHC##8J)
                                                                                                                                                                                                                                                                                                                                              CFEJ = FLCNF*(H5-H1)
                                                                                                                                                                                                                                                                                                                                                                CCMPLTE THE CCNSTRAINT
                                                                                                                                                                      JUJU
                                                                                                                                                                                                                                                                                                                                                   JUU
```

45.

COO

COO

COO

UUU

SOUSSE

```
22C VAL = TSAT (PO)
24C VAL = 1.00
24C VAL = 1.00
24C VAL = 2.00
25C VAL = 2.00
2
```

REST4200 REST42100 REST422100 REST422100

FCRMAT (* CNU=*, E16.4) FCRMAT (* RFD=*, E16.4) FCRMAT (* ONON-NEGATIVITY CONSTRAINT VIOLATEC:*, (7F5.4)) ENC

		00 4 22 00 0 2 00						
		HE O HO OX • H DO		ET				
		TO IL ON VIOLY IN DIL		>				()
		*U & *J & * U + FL		×				OZ.
		W. F XO TX . O NM		C				U
	w	17 4 4 · · · · · · · · · · · · · · · · ·		CLEAR, CCLR, THICK, CTFK, BTSWIN, CTShin, FCRIT, BKPET, CKP				FMBLR, FMCOND, FMFD, FMCIRC, DEXBLR, CEXCNC, CEXFC, CEXCRC HT, CR4, R4 AREAFR
	1	13 • ON 5× 0 1 FN 0		-				C)
	+	HF F 40 00H J 4 U.		2				•
		HOU S ST KO M S LX L		×				L
	S	N AN A O P P AA N		8				u
	FINDS			-				*
	=	HUNU IL ILL SU S - F . N		-				-
	4	T++ + TI NO U W UC -		or.				
	Sur	ZUD O OO XX OO FO F F		U				u
	ANC	TOO > TO OH W TO O		a				2.
	4	54 6 m OA 10 0 0 ms (0# 2		-				×
		AN A AU SE U S WA .		=				û)
	W)	2 4 C W P ZW + 11 -		.5				-
	CONSTANTS	WHO WIN XI III F FE OF IN		0,				-
	2	SOAMS AND OF I THOU ON O		-				4
	-	ט טוע בו ע בי שאר בי בי אאר						E .
	S	OUIM - IN IH & - OLL IM WH		Z				×
	Z	0 -+ ZIT NK NO + - ZD		-				u:
	0	40 . S OO SE T II II O OO OO		3				0
	0	Detail and Se O a AD MAIN MIN		-				()
	>	OIUU +- U + A + U +- FEU Z		a				œ.
	Œ	WOHERSTO THE O WILL IN .		•				-
	Q			×				C
	5	TO COUNTY TO THE TOTAL THE		-		4	~	2.
	Wi .	COM HOW HE W & WILLIAM SE		5		2	()	-
	J	DX Z Z Z * * * > + Z U Z WIN *U -X		-		u	S	C
	w	SALSNED H O H MOTUM NE		×		-	2.	L
	NECESSARY	HEI -DUNCE TA A BUTY -OH A		0		ш		Σ
	***	AT - S - S A S S S S S S S S S S S S S S S	(1)	F		:	36	
	I	ZN Z + + C BUX + D OFONSIUM	S	=		is	S	0
	11E	NONY .OUR . TH BU . FIMANG E .	2			0	2	Z
		1 +TSFT + + KM +FO GNSIGO +3	•	œ		L		C
	Z.	DY SOUTH TO SEE TO SEE SOUT BUILDING	2			u_	5	04
	æ	TON CHIM CO SW UX 1 MUCHT CO GNW	S	3			S	4.6
	Sm	CEMHADIM C+ MOM B + MALHET	2		a.	4	2	***
Z	01	▲~」下○○○○○○○○○○○○○○○○○○○○○○○○○○○○○○○○○○○○		cc				anu
READIN	EAWATER	ANS MAINT F IM TO FEGIND ANS	-	4	NETPAR	EFFFWF, EFFPSW, EFFTUR	MS64, MS65, WS66, WSG7	DOG
2	~ ~	TAT NE STAN TO SECTION	6	u	-	11	3	2 - 0
W	THE W	THE # JOHO COUNTY ONLY WILL SHIP	Z	5	ž	W	E	TIA
ox	MIC	4 LUHUEUH AL ON OX DE DHUEUHUND						
	2	ZHINNING APHITOMMOUNCING OF OZPM	=	=	=	=	=	222
-	==	かいる コールないいけん そういといいかいまいまいかい	2	3	3	20	3	200
SLERCLTINE	-	אבייייי וויים איים ביים בייטייי בייטייי	(5,20) MSG1, MSG2, MSG	(5,30)	(5,40)	(5,50)	(5,60)	55.00 500 500 500 500 500 500 500 500 50
-	OVI	H 00 1 FS 0+ 4 F W3 00	5	5	41	5	10	nna
-	arm.	OZZ> •ZTZZZZ •XZJZJZZTZZZZ##	-	-	-	-	-	
5								
H	VIC	#22 +U2 424 +2U1 +2U2+ EW72222244	REAL	REAL	REAL	PEAC	REAC	FAC
-	w	として でくてしししちなしし しししなをしいしししき	-	-	-	-	-	www
S	via	HOOTTOKO COKODOOKK	Cr.	O.	œ	a.	OC	a.a.a.
	THIS SLENOLTIN							
	-4							

UUUUU

```
nin
                                             = 7.9554-.15215*T1+.0013673*T1**2-4.7842E-06*T1**3
= 30927+5.65E-04*T1
= 30927+5.65E-04*T2
= 1.02645-6.86411E-04*T1+4.0374E-06*T1**2
= 1.02645-6.86411E-04*T1+4.0374E-06*T1**2
= 62.39+.00362111*T2-7.625E-05*T1**2
= 26.173-.54646*T1+005147*T1**2-1.861E-05*T1**3
= 26.173-.54646*T2+.005147*T2**2-1.861E-05*T1**3
                    מנונ
                    AND
                    MARE
                    THE
                    10
(5,40)
                    PROPERTIES
(N.EC.11) REAC
                                                                                                                                                         8877100
8877100
6877100
6877100
                    THE
                                 INE
                                                                                                                                                         ETCRN
                    CETERA
                                                                                                                                                         000000
```

UUUU

```
| SUBSTITUTE | SETA | S
              02
```

UUUU

UUU

UUU

IFFLICIT REAL#8(A-+,0-2,\$)
CCMMON /SHARE/ X(100),DFL(100),A(100,100),N,MN,NF1,NM1
REAL #8KC,NTUC,NTLE,NU,NETPMR,NET,NETGUT
CALL CIFFL (1)
FETURN
FFTURN SUMT TO FINE THE FIRST CERIVATIVE Š THIS SLERCLTINE CALL SLERCLTINE GRADI

UUUU

IAFLICIT REAL*8(A-+,0-Z,\$)
CCMCN /SHARE/ X(100),DEL(100),A(100,100),N,M,MN,NP1,NM1
CALL DIFFZ (J)
FETEN TO FINE THE SECENE CERIVATIVE CA SUMT SLERCUTINE MATRIX (J.L) THIS SLENGLTINE CALL

UUUU

HE COMPUTERS ROLTINE IT THE ALSSELT OCRECIOOOD). SING TO FIND THE COEFFICIENTS FOR FIND NOW EER THROUGH THE TRANSISTION RECICN (200 L CLAP (R,NUL, CNLL, CI, TUBEL, FRSM)
= NUL-AA*R** 8
= DNLL-8*AA*R** (-.2)
= A1/61
= 1.2E0 7*AB+A(1)
= 4.E064 AP (R,NUL, CNUL, CI, TUBEL, FRSh, -AA*R***8 1L-8*AA*R**(-.2) (CI,TLBEL, FRSW) A MATRIX CCMCO /SIM/ A(16), 8(4) SICE VECTOR 182 1812 1808 1808 1808 1804 1804 1804 1804 SET THE VALUES OF THE SLERCLTINE SETA HANC THE RIGHT 2003

COCOC

COO

ANC THE PARTIAL CF IF THE NUSSELT NUMBER IS LESS THAN 3.66 SET IT TO 3.66 AND THE FIRTIAL TO 0.0 NUMBER SLERGUTINE CLAM (R, NUL, CNUL, CI, TLEEL, PRSK) IPFLICIT REAL*8(A-FN-Z,\$)

= 2.02*(PRSW*DI/(fUBEL*12.))**.233333

= A/3.

NLL = A*R**.323333

CNLL = E*R**(-.666667) THIS SUBRCUTINE CALCULATES THE NUSSELT THE NUSSELT CF FLCM. 11 (3.66.LE.(A*12.59921)) GC TO CALL = 3.66 FELLAN G.O FELLAN

00000

5

91

20

LUUU

```
02
```

UUU

UUU

LUU

UUUU

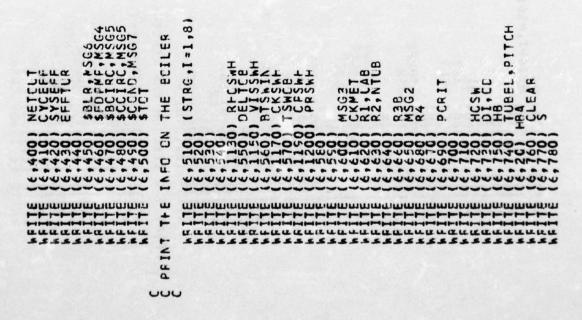
OUU

1F1S

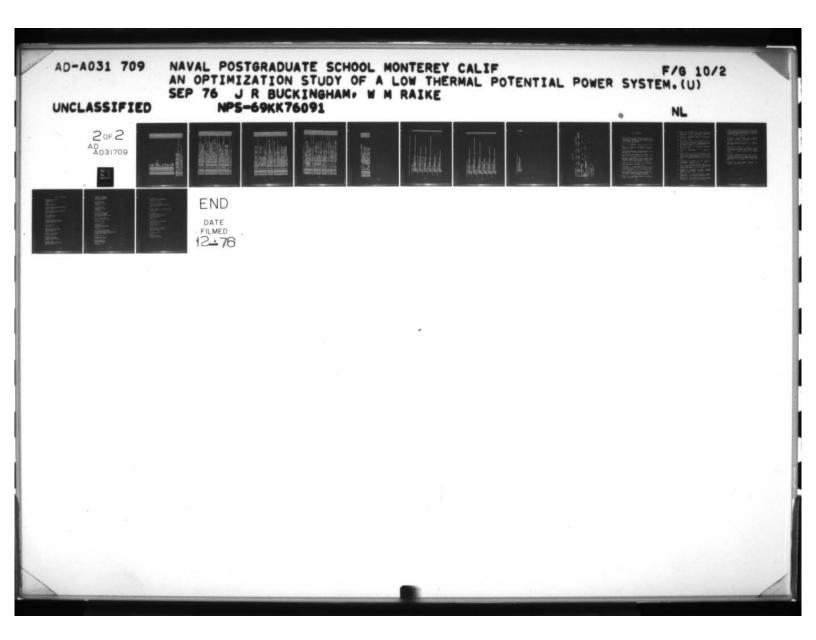
UU

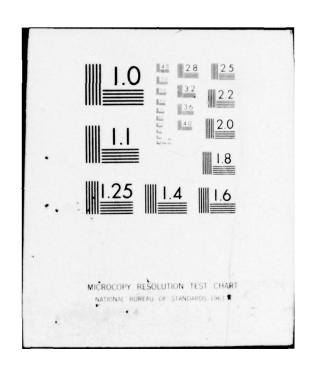
COCOCOC

是 人名伊尔特里 医神经节的 医皮肤的



```
THE CCNCENSER
                                (STRG, I=1,8
DS HEIGHT, MSG1
CSW
         OTUBEE
                                   MONUMN4N
W W W T T
                    Du
   BNI
Z
                                -
արտատարաբարարարարարարարարարարարարարարար
                              ш
                                աաաաաաանաաա
11
                                TARREGERATER OF THE
                                222222222
                              PRIN
                             UUU
```





```
, . + , , 2 A B ,
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                             (PCILER) : HALFY (ETU/LER) : )
L FLCh : )
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                 CAT
                                                                                                                                                                                                                                                                                                                                                                                                                                                                      - 8CN
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                             TENDERATUPE
                                                                                                                             CFITCH
PASAC
                                                                                                                                                                                                CNI
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                   1244
 24444
04040
04040
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
04060
                                                                                                                                                                                                                                                                                                                                                                                                                                               444
0000 0 000
```

```
E FITCH= ; F6.2; (IN).)
                                                     (FT))
                                                 623
                             F12 10)
ARE PRESS AND TEMF CEF')
F12 10)
F12 10)
RANSFER'CCEFFICIENTS'/'+',T62,
        .
                                           F7-31 (FT) BASEC ON
            (BTU/FR-FT-F) ...
        ***
  ---
       E CALCULATIONS
           17.1A8)
17.12.10.5.
 BCILING
                                           TUBE BANK FEIGHT = , 17
       NTERMEDIAT
 THE CHANGE
EAN TO
```

	UTA 2450 UTA 250 UTA 250	でして サンター サン サンター サン サンター サンター サンター サンター サンター サンター サンター サンター サンター サンター サン サンター サンター サンター サンター サンター サンター サンター サンター サンター サンター サン サンター サンター サンター サンター サンター サンター サンター サンター サンター サンター サン サン サン サンター サンター サンター サンター サンター サンター サンター サンター サンター サ		CTAN Services	20000000000000000000000000000000000000	200000 24444 2444 24444 24444 24444 24444 24444 24444 24444 24444 24444 24444 2444 2444 24444 24
CENAT ('+', T62, 'HEAT EXCHANGER AREAS'/+', T62, ' CRMAT ('+', T63, TGTAL PER SHELL (CC) = ', E16.7, (FT2)') CRMAT ('+', T63, TGTAL PER SHELL (CC) = ', E16.7, (FT2)') CRMAT ('O ALLOWABLE VAPOR LOAD=', F5.2' (FT2)') CRMAT ('O BOILER CUTLET = ', F7.2' (FT2)') CRMAT ('O BOILER CUTLET = ', F7.2' (FSIA)') CRMAT ('O BOILER CUTLET = ', F7.2' (FSIA)') CRMAT ('O BOILER CUTLET = ', F7.2' (FSIA)') CRMAT ('I', T37, CONDENSER DATA', '+', 4AE, ', '	CRMAT ('+', T62, INTERMEDIATE CALCULATIONS', /, '+', T62,	FCRNAT (+1762, R1=1712.10, F5.2) (F6.2) (F6.	FCRWAT (++, 162, CARCIOTIVITY= , F9.2) (BTL/FR-FT-F) (CRECATION) (CRE	CRMAT ('+', T62,' PER TUBE = ", F5.2,' (BTL/HR)') CRMAT ('', TUBE LENGTH=', F7.3,' (FT) TUBE PITCH=', F6.3,' (D	CAMAT (++, T62. FER UNIT DRED=', F9.2, '(ETU/HR-FT2)') CAMAT (, T62. FER UNIT DRED=', F6.3, '(IN)') CAMAT (, SPELL S=', F7.2') CAMAT (, T62. HEAT EXCHANGER DRED', '++, T62.'	CRMAT (.+., T62, . TCTAL PER SHELL (CD) = TTEICTTTTFT27;) CRMAT (.+., T62, . TCTAL PER SHELL (CD) = TTEICTTTTTFT27;) CRMAT (.0., SALTMATER VELCCITY=', F8.4,' (FT/SEC)') CRMAT (.0., MASS FLOW RATES')
000000000 0000000000000000000000000000		0000000000000000000000000000000000000	ころろうろろろ	1250	111111 11010101	40000 90000

SALTWATER = 'EIT 8 '(LBM/FR)')

REYNCLOS GUMBERS', 'EIG 6' (LBM/FR)')

TUBE CONTRACTION FACTOR = 'F7.4)

TUBE EXPANSION FACTOR = 'F7.4)

FRICTION FACTOR (FANNING) = 'F9.7)

CONDESSURES '-+

TEMPERATURES /-+

TEMPERATURES /-+

SEAWATER INLET = 'F8.4' (F)')

SEAWATER INLET = 'F8.4' (F)')

```
FLNCTION RECF (P)
IPFLICIT REAL*8(A-+,0-2,$)
RFCF = 3.64527E01-6.12201E-02*P+1.55528E-04*F**2-3.57024E-07*F**3
FTURN
FTURN
FNCF CENSITY
                                                                                                                                                                                                                            FINCTION HG (P)
IMPLICIT REAL#8(A-F.O-Z.$)
FETLRN
FETLRN
                                                                                                                                                               FLNCTICN FF (P)
IMPLICIT REAL*8(A-H,O-Z,$)
FF = 8.51215E01+6.30546E-01*P-1.04803E-03*P**2
FFTLFN
ENC
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                 FLNCTION SLRFT (P)
IPFLICIT REAL+8(A-F,D-Z,$)
SLRFT = 5.5381E-10*(RHCF(P)-RHOG(P))**4
RETLRN
                                                                                                                                                                                                                                                                                                                                                                                                                FLNCTION RECG (P)
IPFLICIT REAL#8(A-F, 0-Z, S)
RFCG = 4.21853E-02+6.75332E-03*P
RFTURN
ENC
SLPFICE TENSION
                                                                                                                                         ENTFALFY OF THE SATURATED LIGUID
                                                                                                                                                                                                                         ENTITLEY OF THE SATURATED VAPOR
SETLEPTION TEMP
                                                                                                                                                                                                                                                                                                         LICLIC DENSITY
                                                                                                                                          UU
                                                                                                                                                                                                                                                                                                          UU
                                                                                                                                                                                                                                                                                                                                                                                           UU
                                                                                                                                                                                                                                                                                                                                                                                                                                                                          UU
```

```
VISCF (T)
REAL*8(A-F,0-Z,$)
(1.08866E01-6.0602E-02*T+2.60276E-04*T**2-1.18555E-06*T**3
                                                                                                                                                                                                                                        FLACTION VISCSV (T)
IPFLICIT REAL+8(A-F,O-Z,$)
VISCSV = (4.78332+1.20747E-O2*T+2.62346E-C5*T**2)*1.E-O6
FRICH
FRICT NUPBER OF PROFANE
                                                                                                                                                                                                                                                                                                                                                                                                                                         FLNCTION ENTF (T)

IPFLICIT REAL+8(A-F-0-2.8)

ENTF = 2.28435E-01+1.17812E-03*T+1.85814E-06*T**2

FETCRN

ENTF SATURATED VAPOR ENTROPY
                           FLACTICA CFF (T)

INFLICIT REAL*8(A-H;0-Z;$)

(FF = 5.62026-01+6.57786E-04*7+2.06525E-06*1**2

RETLEN

ENCLIC'S VISCOSITY
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                          FLNCTION ENTG (1)
10FLICIT REAL*8(A-+,0-2,$)
6NTG = 6.0467E-01-2.2967E-04*T+1.0589E-06*T**2
FILAN
ETCAN
                                                                                                                                                                                                                                                                                                                                          FLACTION PRWF (T)

IPFLICIT REAL*8(A-10-245)

FEFURN

ENTERINE

SATURATED LICLID ENTREFY
C HEAT CAPACITY OF THE LICUIC
                                                                                                                          FLNCTICN VISC
IPFLICIT REAL
VISCE (1.08
1)41.6-05
VAFCR VISCOSITY
                                                                                                                                                                                                                UU
                                                                                                  UU
                                                                                                                                                                                                                                                                                                                 UU
                                                                                                                                                                                                                                                                                                                                                                                                                UU
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                UU
```

THERNAL CONCLCTIVITY OF THE LIGUIC FLACTICITY REAL*8(A-H₀0-Z₃) FELCITY REAL*8(A-H₀0-Z₃) FELCIN GETCHN

UL

. E-054. E 001.8 E C30C1E0C130003	.5 0 00.55 D 002.5 C 002.5 C 00TLRBULEN	.55	0100100100 000000000000000000000000000	1 2 2 2 3 1				
E 001. E-	0 001-17 0	FRCBLEM CUSO-NI 028 -028	00100100	1	E 00	0		
PARAVETER CARC	INITIAL X VECTOR	2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2	821521281641 83223233	FIFE CFTION CARD	TCLERANCE CARC	· SECCNE CPTICN CARD		

LIST OF REFERENCES

- 1. Naval Shir Research and Development Center Report 4195,

 Proceedings of Workshop on Navy Alternate Inergy

 Sources Research and Development, by J.R. Felt and

 H.V. Nutt (eds), January 1974.
- 2. Weyl, F.K., Cceanograpy, John Wiley and Sons, 1970.
- 3. Myers, J.J., Holm, C.H. and McAllister, R.F. (eds), <u>Handlock of Ocean and Underwater Engineering</u>, McGraw-Hill, 1969.
- 4. Second Coean Thremal Energy Conversion Workshop, 26-28 September, 1974. University of Miami, 1974.
- 5. Solar Sea Power Plant Conference and Workshop, 27-28

 June, 1973, Carnegie-Mellon University, 1973.
- 6. Proceedings, Third Workshop on Ocean Thermal Energy Conversion (CTEC), 8-10 May, 1975, The Johns Hopkins University, Applied Physics Laboratory, August 1975.
- 7. Eeveridge, G.S.G. and Schechter, R.S., Optimization:
 Theory and Practice, McGraw-Hill, 1970.
- 8. Starczevski, J., "Generalized Design of Evaporators: Heat Transfer to Nucleate Boiling Liquids", <u>Fritsh</u> <u>Chemical Engineering</u>, v. 10, p. 523-531, August 1965.
- 9. Naval Ecstgraduate School, Report NPS-59Nn75062A, Therececoncaic Analysis of Vapor Power Systems, ty Cdr F.L. Sherrard, Jr., USN, and Others, 30 June 1975.
- 10. Wisher, C.A., Optimization Methods for Large-Scale Systems, McGraw-Hill, 1971.

- 11. Palen, J.W. and Small, W.M., " A New Way to Design Kettle and Internal Reboilers", <u>Hydrocarbon Processing</u>, v. 43, p. 199-208, November 1964.
- 12. Collier, J.G., <u>Convective Boiling and Condensation</u>, McGraw-Hill, 1972.
- 13. Palen, J.W., Yarden, A. and Taborek, J., "
 Characteristics of Boiling Outside Large-Scale
 Horizontal Multitube Bundles", <u>AIChE Symposium Series</u>,
 Heat Transfer Tulsa, v. 68, p. 50-61, 1972.
- 14. Kays, W.M. and London, A.I., <u>Compact Heat Exchangers</u>, McGraw-Fill, 1964.
- 15. Holman, J.P., <u>Heat Transfer</u>, 3rd ed., McGraw-Hill, 1972.
- 16. Mostinski, I.I., " The law of Corresponding States Applied to Calculation of Heat Transfer and Burnouts in Liquid Eciling", <u>Teploenergetika</u>, v. 10, p. 12, 1963, abstract in English in Erit. Chem. Eng., v.8, p.580, 1963.
- 17. Rohsenow, W.M. and Hartnett, J.P. (eds), <u>Handbook of</u>
 <u>Heat transfer</u>, McGraw-Hill, 1973.
- 18. Chen, M.M., " Analytical Film Condensation: Fart 2 Single and Multiple Horizontal Tubes", <u>Journal of Heat Iransfer</u>, <u>Trais</u>. <u>ASME</u>, v. 83, p. 55, 1961.
- 19. Eaumeister, T. and Marks, L.S. (eds), <u>Standard</u>

 <u>Handbock for Mechanical Engineers</u>, 7th ed.,

 McGraw-Fill, 1967.
- 20. Guthrie, K.M., " Capital Cost Estimating", Chemical Engineering, v. 76, p. 114-142, 24 March 1969.
- 21. Guthrie, K.E., " Pump and Valve Costs", Chemical Engineering, v. 78, p. 151-159, 11 October 1971.

- 22. Research Analsis Corporation Report BAC-P-63, A Guide to SUMT Version 4: The Computer Program Implementing the Sequential Unconstrained Minimization Technique for Nonlinear Programming, by W.C. Mylander, R.L. Bolmes, and G.F. McCormick, October 1971.
- 23. Fiacco, A.V. and McCormick, G.F., Nonlinear Programming: Sequential Unconstrained Minimization Techniques, John Wiley and Sons, 1968.
- 24. Duffin, R.J., Peterson, E.I. and Zener, C., <u>Geometric</u>

 <u>Programming: Theory and Application</u>, John Wiley and Scns, 1967.
- 25. Carnegie-Mellon University, Technical Feport NSF/RANN/SE/GI-39114/PR/73/1, First Cuarterly Frogress Report Covering the Period June 1,1973 to October 1,1573, by C. Zener, A. Lavi and C.C. Wu, October 11, 1973.
- 26. University of Massachusetts (Amherst), Technical Report NSF/RANN/SE/GI-34979/TR/75/5, Design and Off-Lesign performance Analysis of Ocean Thermal Difference Power Flant Turbines, by R.J. Veenema and I.L. Ambs, May 1975.
- 27. Bartlett, R.I., <u>Steam Turbine Ferformance and Economics</u>, McGraw-Hill, 1958.

INITIAL DISTRIBUTION LIST

1.	Defense Documentation Center Cameron Station	
	Alexandria, VA 22314	12
2.	Library Naval Postgraduate School Monterey, CA 93940	2
3.	Chairman, Department of Mechanical Engineering Naval Postgraduate School Monterey, CA 93940	1
4.	Chairman, Department of Operations Research Naval Postgraduate School Monterey, CA 93940	1
5.	Dr. R. H. Nunn Office of Naval Research Branch Office, London P. O. Box 39 FPO New York 09510	1
6.	Associate Professor M. D. Kelleher Department of Mechanical Engineering Naval Postgraduate School Monterey, CA 93940	5
7.	LCDR J. R. Buckingham USS AJAX AR-6 FPO San Francisco, CA 96602	1
8.	Associate Professor W. M. Raike Department of Operations Research Naval Postgraduate School Monterey, CA 93940	5
9.	Commander F. L. Sheppard, Jr. Mare Island Naval Shipyard Vallejo, CA 94592	1
LO.	Dr. William E. Heronemus University of Massachusetts Department of Civil Engineering Amherst, MA 01002	1
11.	Dr. Jon G. McGowan Department of Mechanical Engineering University of Massachusetts Amherst. MA 01002	

12.	Professor E. C. Haderlie Department of Oceanography Naval Postgraduate School Monterey, CA 93940	1
13.	Dr. Clarence Zener University Professor Carnegie-Mellon University Schenley Park Pittsburgh, PA 15213	1
14.	Dr. John Fetkovich Department of Physics Carnegie-Mellon University Schenley Park Pittsburgh, PA 15213	1
15.	Energy Program Office, Code L80 Attn: Mr. M. J. Slaminski Civil Engineering Laboratory Naval Construction Battalion Center Port Hueneme, CA 73043	1
16.	Dr. G. W. Leonard, Head Propulsion Development Department Code 45 Naval Weapons Center China Lake, CA 93555	1
17.	Dr. Arthur L. Austin Project Leader Geothermal Development Program Lawrence Livermore Laboratory P.O. Box 808 Livermore, CA 94551	
18.		1
19.	Dr. Owen M. Griffin Code 8441 Naval Research Laboratory Washington, D.C. 20375	1
20.	Mr. John B. Gregory Ocean Technology Program Office of Naval Research Code 485	1

21.	Dr. Abraham Lavi Department of Electrical Engineering Carnegie-Mellon University Pittsburgh, PA 15213	1
22.	Commander Paul A. Petzrich (C.E.C.) U.S.N. Director U.S. Navy Energy and Natural Resources R&D NSRDC, Annapolis Laboratory Annapolis, MD 21402	1
23.	Dr. Earl R. Quandt U.S. Naval Ship Research and Development Center Annapolis Division Bethesda, MD 20034	1
24.	Dr. H. Tabor Scientific Director The Scientific Research Foundation Dan Danciger Building Hebrew University Campus P.O.B. 3745 Jerusalem, ISRAEL	1
25.	Dr. Lloyd Lewis Naval Facilities Engineering Command Chesapeake Division Naval Station, Washington Naval Yard Annex Ace and M Streets S.E. Washington, D.C. 20374	1
26.	Associate Professor J. K. Hartman Department of Operations Research Naval Postgraduate School Monterey, CA 93940	
27.	CAPT Vincent M. Skrinak Code NAVMAT 03Z 824 CP6 Department of the Navy 2221 Jefferson-Davis Hwy Arlington, Virginia 20360	1